

Dynamic Droop Control in Low-Inertia Power Systems

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Abstract-A widely embraced approach to mitigate the dynamic degradation in low-inertia power systems is to mimic generation response using grid-connected inverters to restore the stiffness of the grid. In this article, we seek to challenge this approach and advocate for a principled design based on a systematic analysis of the performance trade-offs of inverter-based frequency control. With this aim, we perform a qualitative and quantitative study comparing the effect of conventional control strategies-droop control (DC) and virtual inertia (VI)-on several performance metrics induced by \mathcal{L}_{2} and \mathcal{L}_{∞} signal norms. By extending a recently proposed modal decomposition method, we capture the effect of step and stochastic power disturbances, and frequency measurement noise, on the overall transient and steady-state behavior of the system. Our analysis unveils several limitations of these solutions, such as the inability of DC to improve dynamic frequency response without increasing steady-state control effort, or the large frequency variance that VI introduces in the presence of measurement noise. We further propose a novel dynam-i-c droop controller (iDroop) that overcomes the limitations of DC and VI. More precisely, we show that iDroop can be tuned to achieve high noise rejection, fast system-wide synchronization, or frequency overshoot (Nadir) elimination without affecting the steady-state control effort share, and propose a tuning recommendation that strikes a balance among these objectives. Extensive numerical experimentation shows that the proposed tuning is effective even when our proportionality assumption is not valid, and that the particular tuning used for Nadir elimination strikes a good trade-off among various performance metrics.

Index Terms—Frequency control, low-inertia power systems, static and dynamic performance.

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I. INTRODUCTION

T HE shift from conventional synchronous generation to renewable converter-based sources has recently led to a noticeable degradation of the power system frequency dynamics [1]. At the center of this problem is the reduction of the systemwide inertia that accentuates frequency fluctuations in response to disturbances [2], [3]. Besides increasing the risk of frequency instabilities and blackouts [4], this dynamic degradation also places limits on the total amount of renewable generation that can be sustained by the grid [5]. Ireland, for instance, is already resorting to wind curtailment whenever wind becomes larger than 50% of existing demand in order to preserve the grid stability [6].

A widely embraced approach to mitigate this problem is to mimic synchronous generation response using grid-connected converters [7]; that is, to introduce *virtual inertia* to restore the stiffness that the system used to enjoy [8]. Notable works within this line of research focus on leveraging computational methods [9]-[11] to efficiently allocate synthetic inertial or droop response, or analytical methods that characterize the sensitivity of different performance metrics to global or spatial variations of system parameters [12]–[14]. However, to this day, it is unclear whether this particular choice of control is the most suitable for the task. On the one hand, unlike synchronous generators that leverage stored kinetic energy to modulate electric power injection, converter-based controllers need to actively change their power injection based on noisy measurements of frequency or power. On the other hand, converter-based control can be significantly faster than conventional generators. Therefore, using converters to mimic generator behavior does not take advantage of their full potential. In this article, we seek to challenge this approach of mimicking generation response and advocate for a principled control design perspective.

To achieve this goal, we build on recent efforts by the control community on quantifying power network dynamic performance using \mathcal{L}_2 and \mathcal{L}_∞ norms [9], [15], and perform a systematic study evaluating the effect of different control strategies, such as droop control (DC) [16] and virtual inertia (VI) [17], on a set of static and dynamic figures of merits that are practically relevant from the power engineering standpoint. More precisely, under a mild—yet insightful—proportionality assumption, we compute closed-form solutions and sensitivities of controller parameters on the steady-state control effort share, frequency Nadir, \mathcal{L}_2 -synchronization cost, and frequency variance of the response of a power network to step and stochastic disturbances.

0018-9286 © 2020 IEEE. Personal use is permitted, but republication/redistribution requires IEEE permission. See https://www.ieee.org/publications/rights/index.html for more information. Our analysis unveils the inability of DC and VI to cope with seemingly opposing objectives, such as synchronization cost reduction without increasing steady-state effort share (DC), or frequency Nadir reduction without high frequency variance (VI). Therefore, rather than clinging to the idea of efficiently allocating synthetic inertia or droop, we advocate the search of a better solution.

To this end, we propose novel dynamic droop (iDroop) control—inspired by classical lead/lag compensation—which outperforms current control strategies (VI and DC) in an overall sense. More precisely, the following points are provided.

- Unlike DC that sacrifices steady-state effort share to improve dynamic performance, the added degrees of iDroop allow to decouple steady-state effort from dynamic performance improvement.
- Unlike VI that amplifies frequency measurement noise, the lead/lag property of iDroop makes it less sensitive to noise and power disturbances, as measured by the \mathcal{H}_2 norm [18] of the input–output system defined from measurement noise and power fluctuations to frequency deviations.
- iDroop can further be tuned to either eliminate the frequency Nadir (i.e., remove the frequency overshoot), by compensating for the turbine lag, or to eliminate synchronization cost; a feature shown to be unattainable by VI.

All of the above properties are attained through rigorous analysis on explicit expressions for performance metrics that are achieved under a mild yet insightful proportionality assumption that generalizes prior work [1], [2].

We further validate our analysis through extensive numerical simulations, performed on a low-inertia system—the Icelandic Grid—that does not satisfy our parameter assumptions. Our numerical results also show that iDroop with the Nadir-eliminated tuning designed based on the proportional parameters assumption works well even in environments with mixed step and stochastic disturbances.

The rest of this article is organized as follows. Section II describes the power network model and defines performance metrics. Section III introduces our assumptions and a system diagonalization that eases the computations and derives some generic results that provide a foundation for further performance analysis. Section IV analyzes both steady-state and dynamic performance of DC and VI, illustrates their limitations, and motivates the need for a new control strategy. Section V describes the proposed iDroop and shows how it outperforms DC and VI from different perspectives. Section VI validates our results through detailed simulations. Section VII concludes the article.

II. PRELIMINARIES

A. Power System Model

We consider a connected power network composed of n buses indexed by $i \in \mathcal{V} := \{1, \ldots, n\}$ and transmission lines denoted by unordered pairs $\{i, j\} \in \mathcal{E}$, where \mathcal{E} is a set of 2-element subsets of \mathcal{V} . As illustrated by the block diagram in Fig. 1, the system dynamics are modeled as a feedback interconnection of bus dynamics and network dynamics. The input signals



Fig. 1. Block diagram of power network.

 $p_{\text{in}} := (p_{\text{in},i}, i \in \mathcal{V}) \in \mathbb{R}^n$ and $d_p := (d_{p,i}, i \in \mathcal{V}) \in \mathbb{R}^n$ represent power injection set point changes and power fluctuations around the set point, respectively, and $n_\omega := (n_{\omega,i}, i \in \mathcal{V}) \in \mathbb{R}^n$ represents frequency measurement noise. The weighting functions $\hat{W}_p(s)$ and $\hat{W}_{\omega}(s)$ can be used to adjust the magnitude of these disturbances in the usual way. The output signal $\omega := (\omega_i, i \in \mathcal{V}) \in \mathbb{R}^n$ represents the bus frequency deviation from its nominal value. We now discuss the dynamic elements in more detail.

1) Bus Dynamics: The bus dynamics that maps the net power bus imbalance $u_P := (u_{P,i}, i \in \mathcal{V}) \in \mathbb{R}^n$ to the vector of frequency deviations ω can be described as a feedback loop that comprises a forward-path $\hat{G}(s)$ and a feedback-path $\hat{C}(s)$, where $\hat{G}(s) := \text{diag}(\hat{g}_i(s), i \in \mathcal{V})$ and $\hat{C}(s) := \text{diag}(\hat{c}_i(s), i \in \mathcal{V})$ are the transfer function matrices of generators and inverters, respectively.

a) Generator dynamics: The generator dynamics are composed of the standard swing equations with a turbine, i.e.,

$$m_i \dot{\omega}_i = -d_i \omega_i + q_{\mathrm{r},i} + q_{\mathrm{t},i} + u_{\mathrm{P},i} \tag{1}$$

where $m_i > 0$ denotes the aggregate generator inertia, $d_i > 0$ the aggregate generator damping, $q_{r,i}$ the controllable input power produced by the grid-connected inverter, and $q_{t,i}$ the change in the mechanical power output of the turbine. The turbine does not react to the frequency deviation ω_i until it exceeds a preset threshold $\omega_{\epsilon} \ge 0$, i.e.,

with

$$\tau_i \dot{q}_{t,i} = \varphi_{\omega_{\epsilon}}(\omega_i) - q_{t,i} \tag{2}$$

$$\varphi_{\omega_{\epsilon}}(\omega_{i}) := \begin{cases} -r_{\mathbf{t},i}^{-1}(\omega_{i} + \omega_{\epsilon}) & \omega_{i} \leq -\omega_{\epsilon} \\ 0 & -\omega_{\epsilon} < \omega_{i} < \omega_{\epsilon} \\ -r_{\mathbf{t},i}^{-1}(\omega_{i} - \omega_{\epsilon}) & \omega_{i} \geq \omega_{\epsilon} \end{cases}$$

where $\tau_i > 0$ represents the turbine time constant and $r_{t,i} > 0$ the turbine droop coefficient.

Two special cases of our interest are as follows.

Generator Dynamics 1. (Standard swing dynamics): When $|\omega_i(t)| < \omega_{\epsilon}$, the turbines are not triggered and the generator

dynamics can be described by the transfer function

$$\hat{g}_i(s) = \frac{1}{m_i s + d_i} \tag{3}$$

which is exactly the standard swing dynamics.

Generator Dynamics 2. (Second-order turbine dynamics): When $\omega_{\epsilon} = 0$, the turbines are constantly triggered and the generator dynamics can be described by the transfer function

$$\hat{g}_i(s) = \frac{\tau_i s + 1}{m_i \tau_i s^2 + (m_i + d_i \tau_i) s + d_i + r_{\mathrm{t},i}^{-1}} \,. \tag{4}$$

b) Inverter dynamics: We consider the most commonly used type of inverters, i.e., the grid-following inverters, which adjust their power injection to the network in response to frequency deviations. Since power electronics are significantly faster than the electro-mechanical dynamics of generators, we assume that each inverter measures the local grid frequency deviation ω_i and instantaneously updates the output power $q_{r,i}$. Different control laws can be used to map ω_i to $q_{r,i}$. We represent such laws using a transfer function $\hat{c}_i(s)$. The two most common ones are as follows.

Inverter Dynamics 1. (Droop Control): This control law can provide additional droop capabilities and is given by

$$\hat{c}_i(s) = -r_{\mathrm{r},i}^{-1}$$
 (5)

where $r_{r,i} > 0$ is the droop coefficient.

Inverter Dynamics 2. (Virtual Inertia): Besides providing additional droop capabilities, this control law can compensate the loss of inertia and is given by

$$\hat{c}_i(s) = -\left(m_{\rm v,i}s + r_{\rm r,i}^{-1}\right) \tag{6}$$

where $m_{v,i} > 0$ is the virtual inertia constant.

2) Network Dynamics: The network power fluctuations $p_e := (p_{e,i}, i \in \mathcal{V}) \in \mathbb{R}^n$ are given by a linearized model of the power flow equations [19]:

$$\hat{p}_{\rm e}(s) = \frac{L_{\rm B}}{s} \hat{\omega}(s) \tag{7}$$

where $\hat{p}_{e}(s)$ and $\hat{\omega}(s)$ denote the Laplace transforms of p_{e} and ω , respectively.¹ The matrix L_{B} is an undirected weighted Laplacian matrix of the network with elements

$$L_{\mathrm{B},ij} = \partial_{\theta_j} \sum_{j=1}^n |V_i| |V_j| b_{ij} \sin(\theta_i - \theta_j) \Big|_{\theta = \theta_0}.$$

Here, $\theta := (\theta_i, i \in \mathcal{V}) \in \mathbb{R}^n$ denotes the angle deviation from its nominal, $\theta_0 := (\theta_{0,i}, i \in \mathcal{V}) \in \mathbb{R}^n$ are the equilibrium angles, $|V_i|$ is the (constant) voltage magnitude at bus *i*, and b_{ij} is the line $\{i, j\}$ susceptance.

3) Closed-Loop Dynamics: We will investigate the closedloop responses of the system in Fig. 1 from the power injection set point changes p_{in} , the power fluctuations around the set point d_p , and frequency measurement noise n_{ω} to frequency deviations ω , which can be described compactly by the transfer function matrix

$$\hat{T}(s) := \begin{bmatrix} \hat{T}_{\omega \mathrm{p}}(s) & \hat{T}_{\omega \mathrm{dn}}(s) := \begin{bmatrix} \hat{T}_{\omega \mathrm{d}}(s) & \hat{T}_{\omega \mathrm{n}}(s) \end{bmatrix} \end{bmatrix}.$$
(8)

Remark 1. (Model assumptions): The *linearized* network model (8) implicitly makes the following assumptions which are

standard and well-justified for frequency control on transmission networks [20].

- Bus voltage magnitudes $|V_i|$'s are constant; we are not modeling the dynamics of exciters used for voltage control; these are assumed to operate at a much faster time-scale.
- Lines {*i*, *j*} are lossless.
- Reactive power flows do not affect bus voltage phase angles and frequencies.
- Without loss of generality, the equilibrium angle difference (θ_{0,i} θ_{0,j}) across each line is less than π/2.

For a first principle derivation of the model, we refer to [21, Section VII]. For applications of similar models for frequency control within the control literature (see, e.g., [22]–[24]).

Remark 2. (Internal stability of (8)): Throughout this article, we consider feedback interconnections of positive real and strictly positive real subsystems. Internal stability follows from classical results [25]. Since the focus of this article is on performance, we do not discuss internal stability here in detail. We refer to the reader to [26], for a thorough treatment of similar feedback interconnections. From now on, a standing assumption—that can be verified—is that feedback interconnection described in Fig. 1 is internally stable.

Remark 3. (Model extensions and limitations): Our model accepts generalizations accounting for constant power loads via Kron reduction [20] and lossy lines with uniform R/X ratios [27], which are not presented due to space constraints. Also, we would like to point out that constant power loads are good surrogates to resistive loads, since the demand variations of resistive loads are mostly driven by voltage fluctuations which are ignorable in the context of frequency control on transmission networks. Admittedly, our model cannot capture frequency-dependent loads or lossy lines with heterogeneous R/X ratios, which are possible directions of future research.

B. Performance Metrics

Having considered the model of the power network, we are now ready to introduce performance metrics used in this article to compare different inverter control laws.

 Steady-State Effort Share: This metric measures the fraction of the power imbalance addressed by inverters, which is calculated as the absolute value of the ratio between the inverter steady-state output power and the total power imbalance, i.e.,

$$\mathrm{ES} := \left| \frac{\sum_{i=1}^{n} \hat{c}_i(0) \omega_{\mathrm{ss},i}}{\sum_{i=1}^{n} p_{\mathrm{in},i}(0^+)} \right| \tag{9}$$

when the system $\hat{T}_{\omega_{\rm P}}$ undergoes a step change in power excitation. Here, $\hat{c}_i(0)$ is the dc gain of the inverter and $\omega_{{\rm ss},i}$ is the steady-state frequency deviation. A higher steady-state effort share indicates that a larger amount of steady-state power output is required from inverters in the process of handling certain power imbalance. Since this necessary headroom is usually achieved by either curtailment or additional storage [28], [29], a lower steady-state effort can be associated with lower operational costs and it is therefore desired.

¹We use hat to distinguish the Laplace transform from its time domain counterpart.

2) Power Fluctuations and Measurement Noise: This metric measures how the relative intensity of power fluctuations and measurement noise affect the frequency deviations, as quantified by the \mathcal{H}_2 norm of the transfer function $\hat{T}_{\omega dn}$

$$\|T_{\omega dn}\|_{\mathcal{H}_2}^2$$

$$:= \begin{cases} \frac{1}{2\pi} \int_{-\infty}^{\infty} \operatorname{tr}(\mathrm{T}_{\omega \mathrm{dn}}(j\omega)^* \mathrm{T}_{\omega \mathrm{dn}}(j\omega)) \mathrm{d}\omega & \text{if } T_{\omega \mathrm{dn}} \text{ is stable} \\ \infty & \text{otherwise}^2. \end{cases}$$
(10)

The quantity $\|\hat{T}_{\omega dn}\|_{\mathcal{H}_2}$ has several standard interpretations in terms of the input–output behavior of the system $\hat{T}_{\omega dn}$ [18]. In particular, in the stochastic setting, when the disturbance signals $d_{\mathrm{p},i}$ and $n_{\omega,i}$ are independent, zero mean, unit variance, white noise, then $\lim_{t\to\infty} \mathbb{E}[\omega(t)^T \omega(t)] = \|\hat{T}_{\omega dn}\|_{\mathcal{H}_2}^2$.

This means that the sum of the steady-state variances in the output of $\hat{T}_{\omega dn}$ in response to these disturbance equals the squared \mathcal{H}_2 norm of $\hat{T}_{\omega dn}$. Thus, the \mathcal{H}_2 norm gives a precise measure of how the intensity of power fluctuations and measurement noise affects the frequency deviations of the system.

3) Synchronization Cost: This metric measures the size of individual bus deviations from the synchronous response when the system $\hat{T}_{\omega p}$ is subject to a step change in power excitation given by $p_{in} = u_0 \mathbb{1}_{t \ge 0} \in \mathbb{R}^n$, where $u_0 \in \mathbb{R}^n$ is a given vector direction that allows for power disturbances of different magnitudes at individual nodes and $\mathbb{1}_{t \ge 0}$ is the unit-step function [15]. This is quantified by the squared \mathcal{L}_2 norm of the vector of deviations $\tilde{\omega} := \omega - \bar{\omega} \mathbb{1}_n \in \mathbb{R}^n$, i.e.,

$$\|\tilde{\omega}\|_{2}^{2} := \sum_{i=1}^{n} \int_{0}^{\infty} \tilde{\omega}_{i}(t)^{2} \mathrm{d}t .$$
 (11)

Here, $\bar{\omega} := (\sum_{i=1}^{n} m_i \omega_i) / (\sum_{i=1}^{n} m_i)$ is the system frequency that corresponds to the inertia-weighted average of bus frequency deviations and $\mathbb{1}_n \in \mathbb{R}^n$ is the vector of all ones.

4) Nadir: This metric measures the minimum postcontingency frequency of a power system, which can be quantified by the \mathcal{L}_{∞} norm of the system frequency $\bar{\omega}$, i.e.,

$$\|\bar{\omega}\|_{\infty} := \max_{t \ge 0} |\bar{\omega}(t)| \tag{12}$$

when the system $T_{\omega_{\rm P}}$ has as input a step change in power excitation [15], i.e., $p_{\rm in} = u_0 \mathbb{1}_{t \ge 0} \in \mathbb{R}^n$. This quantity matters in that deeper Nadir increases the risk of underfrequency load shedding and cascading outrages.

III. RESULTS

In this section, we show that under a simplifying assumption, it is possible to compute all performance metrics introduced in Section II-B analytically as functions of the system parameters, which pave us a way to formally compare the conventional control laws DC and VI in Section IV as well as suggest an improved control law iDroop in Section V. We remark that the assumptions are only used in the analysis, but as shown in Section VI, the insights and advantages of the proposed solution are still there when these assumptions do not hold.

 ^{2}j represents the imaginary unit which satisfies $j^{2}=-1$ and ω represents the frequency variable.



Fig. 2. Equivalent block diagrams of power network under proportionality assumption.

A. Diagonalization

In order to make the analysis tractable, we require the closedloop transfer functions to be diagonalizable. This is ensured by the following assumption, which is a generalization of [13], [15].

Assumption 1. (Proportionality): There exists a proportionality matrix $F := \text{diag}(f_i, i \in \mathcal{V}) \in \mathbb{R}_{\geq 0}^{n \times n}$ such that

$$\hat{G}(s) = \hat{g}_{\mathrm{o}}(s)F^{-1}$$
 and $\hat{C}(s) = \hat{c}_{\mathrm{o}}(s)F$

where $\hat{g}_{o}(s)$ and $\hat{c}_{o}(s)$ are called the representative generator and the representative inverter, respectively.

Remark 4. (Proportionality parameters): The parameters f_i 's represent the individual machine rating. This definition is rather arbitrary for our analysis, provided that Assumption 1 is satisfied. Other alternatives could include $f_i = m_i$ or $f_i = m_i/m$ where m is, for example, either the average or maximum generator inertia. The practical relevance of Assumption 1 is justified, for example, by the empirical values reported in [30], which show that, at least in regards of order of magnitude, Assumption 1 is a reasonable first-cut approximation to heterogeneity.

Under Assumption 1, the representative generators of (3) and (4) are given by

 $\hat{g}_{\rm o}(s) = \frac{1}{ms + d}$

and

$$(s) = \frac{\tau s + 1}{m\tau s^2 + (m + d\tau) s + d + r_{\rm t}^{-1}}$$
(14)

respectively, with $m_i = f_i m$, $d_i = f_i d$, $r_{\mathrm{t},i} = r_{\mathrm{t}}/f_i$, and $\tau_i = \tau$.³

Similarly, the representative inverters of DC (5) and VI (6) are given by

$$\hat{c}_{\rm o}(s) = -r_{\rm r}^{-1}$$
 (15)

and

$$\hat{c}_{\rm o}(s) = -\left(m_{\rm v}s + r_{\rm r}^{-1}\right)$$
 (16)

with $m_{v,i} = f_i m_v$ and $r_{r,i} = r_r/f_i$. Using Assumption 1, we can derive a diagonalized version

of (8). First, we rewrite $\hat{G}(s) = F^{-\frac{1}{2}}[\hat{g}_{o}(s)I_{n}]F^{-\frac{1}{2}}$ and $\hat{C}(s) = F^{\frac{1}{2}}[\hat{c}_{o}(s)I_{n}]F^{\frac{1}{2}}$, and after a loop transformation obtain Fig. 2(a). Then, we define the scaled Laplacian matrix

$$L_{\rm F} := F^{-\frac{1}{2}} L_{\rm B} F^{-\frac{1}{2}} \tag{17}$$

(13)

³We use variables without subscript i to denote parameters of the representative generator and inverter.



Fig. 3. Diagonalized block diagram of power network.

by grouping the terms in the upper block of Fig. 2(a). Moreover, since $L_{\rm F} \in \mathbb{R}^{n \times n}$ is symmetric positive semidefinite, it is real orthogonally diagonalizable with non-negative eigenvalues [31]. Thus, there exists an orthogonal matrix $V \in \mathbb{R}^{n \times n}$ with $V^T V = VV^T = I_n$, such that

$$L_{\rm F} = V \Lambda V^T \tag{18}$$

where $\Lambda := \operatorname{diag}(\lambda_k, k \in \{1, \ldots, n\}) \in \mathbb{R}_{\geq 0}^{n \times n}$ with λ_k being the *k*th eigenvalue of L_{F} ordered non-decreasingly $(0 = \lambda_1 < \lambda_2 \leq \cdots \leq \lambda_n)^4$ and $V := \left[(\sum_{i=1}^n f_i)^{-\frac{1}{2}} F^{\frac{1}{2}} \mathbb{1}_n \quad V_\perp \right]$ with $V_\perp := \begin{bmatrix} v_2 \quad \cdots \quad v_n \end{bmatrix}$ composed by the eigenvector v_k associated with λ_k .⁵ Now, applying (17) and (18) to Fig. 2(a) and rearranging blocks of V and V^T results in Fig. 2(b). Finally, moving the block of $\hat{c}_0(s)I_n$ ahead of the summing junction and combining the two parallel paths produces Fig. 3, where the boxed part is fully diagonalized.

Now, by defining the closed-loop with a forward-path $\hat{g}_0(s)I_n$ and a feedback-path $(\Lambda/s - \hat{c}_0(s)I_n)$ as

$$\hat{H}_{\mathrm{p}}(s) = \mathrm{diag}\left(\hat{h}_{\mathrm{p},k}(s), k \in \{1, \dots, n\}\right)$$

where

$$\hat{h}_{\mathrm{p},k}(s) = \frac{\hat{g}_{\mathrm{o}}(s)}{1 + \hat{g}_{\mathrm{o}}(s) \left(\lambda_k/s - \hat{c}_{\mathrm{o}}(s)\right)}$$
(19)

and $\hat{H}_{\omega}(s) = \hat{c}_{\mathrm{o}}(s)\hat{H}_{\mathrm{p}}(s)$, i.e.,

$$\omega(s) = \operatorname{diag}\left(\hat{h}_{\omega,k}(s), k \in \{1, \dots, n\}\right)$$

where

$$h_{\omega,k}(s) = \hat{c}_{\mathrm{o}}(s)h_{\mathrm{p},k}(s) \tag{20}$$

the closed-loop transfer functions from $p_{\rm in}, d_{\rm p}$, and n_{ω} to ω become

$$\hat{T}_{\omega p}(s) = F^{-\frac{1}{2}} V \hat{H}_{p}(s) V^{T} F^{-\frac{1}{2}}$$
(21a)

$$\hat{T}_{\omega d}(s) = F^{-\frac{1}{2}} V \hat{H}_{p}(s) V^{T} F^{-\frac{1}{2}} \hat{W}_{p}(s)$$
(21b)

$${}_{n}(s) = F^{-\frac{1}{2}} V \hat{H}_{\omega}(s) V^{T} F^{\frac{1}{2}} \hat{W}_{\omega}(s)$$
(21c)

respectively.

 \hat{T}_{ω}

Note that depending on the specific generator and inverter dynamics involved, we may add subscripts in the name of a transfer function without making a further declaration in the rest of this article. For example, we may add "T" if the turbine is triggered and "DC" if the inverter operates in DC mode as in $\hat{h}_{\mathrm{p,k,T,DC}}(s)$.

B. Generic Results for Performance Metrics

We now derive some important building blocks required for the performance analysis of the system \hat{T} described in (21). As described in Section II-B, the sensitivity to power fluctuations and measurement noise can be evaluated through the \mathcal{H}_2 norm of the system $\hat{T}_{\omega dn}$, while the steady-state effort share, synchronization cost, and Nadir can all be characterized by a step response of the system $\hat{T}_{\omega p}$. There are two scenarios that are of our interest.

Assumption 2. (Proportional weighting scenario):

The noise weighting functions are given by

$$\hat{W}_{\mathrm{p}}(s) = \kappa_{\mathrm{p}} F^{\frac{1}{2}}$$
 and $\hat{W}_{\omega}(s) = \kappa_{\omega} F^{-\frac{1}{2}}$

where $\kappa_{\rm p} > 0$ and $\kappa_{\omega} > 0$ are weighting constants.

 |ω_i(t)| < ω_ε, ∀i ∈ V and t ≥ 0 such that turbines will not be triggered.

Assumption 3. (Step input scenario):

- There is a step change as defined in Section II-B on the power injection set point, i.e., p_{in} = u₀1_{t≥0}, d_p = 0_n, and n_ω = 0_n with 0_n ∈ ℝⁿ being the vector of all zeros.
- $\omega_{\epsilon} = 0$ such that turbines are constantly triggered.

Remark 5. (Weighting assumption): As a natural counterpart of Assumption 1, we look at the case when the power fluctuations and measurement noise are weighted directly and inversely proportional to the square root of the bus ratings, respectively. In the case of $\hat{W}_p(s)$, this is equivalent to assuming that demand fluctuation variances are proportional to the bus ratings, which is in agreement with the central limit theorem. For $\hat{W}_{\omega}(s)$, this is equivalent to assume noise variances are inversely proportional to the bus ratings, which is in line with the inverse relationship between jitter variance and power consumption for an oscillator in phase-locked-loop [32].

1) Steady-State Effort Share: As indicated by (9), the key of computing the steady-state effort share lies in computing the steady-state frequency deviation ω_{ss} of the system $\hat{T}_{\omega p}$. When the system synchronizes, the steady-state frequency deviation is given by $\omega_{ss} = \omega_{syn} \mathbb{1}_n$ and ω_{syn} is called the synchronous frequency. In the absence of a secondary control layer, e.g., automatic generation control [33], the system can synchronize with a nontrivial frequency deviation, i.e., $\omega_{syn} \neq 0$.

The following lemma provides a general expression for $\omega_{\rm syn}$ in our setting.

Lemma 1. (Synchronous frequency): Let Assumption 3 hold. If $q_{r,i}$ is determined by a control law $\hat{c}_i(s)$, then the output ω of the system $\hat{T}_{\omega p}$ synchronizes to the steady-state frequency deviation $\omega_{ss} = \omega_{syn} \mathbb{1}_n$ with

$$\omega_{\rm syn} = \frac{\sum_{i=1}^{n} u_{0,i}}{\sum_{i=1}^{n} \left(d_i + r_{\rm t,i}^{-1} - \hat{c}_i(0) \right)} .$$
(22)

Proof: See Appendix A.

Now, the theorem below provides an explicit expression for the steady-state effort share.

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⁴Recall that we assume the power network is connected, which means that $L_{\rm F}$ has a single eigenvalue at the origin.

⁵We use k and l to index dynamic modes but i and j to index bus numbers.

Theorem 1. (Steady-state effort share): Let Assumption 3 hold. If $q_{r,i}$ is determined by a control law $\hat{c}_i(s)$, then the steady-state effort share of the system $\hat{T}_{\omega D}$ is given by

$$ES = \left| \frac{\sum_{i=1}^{n} \hat{c}_i(0)}{\sum_{i=1}^{n} \left(d_i + r_{t,i}^{-1} - \hat{c}_i(0) \right)} \right|.$$
 (23)

Proof: It follows directly from Lemma 1 that $\omega_{ss,i} = \omega_{syn} \operatorname{and} \sum_{i=1}^{n} u_{0,i} = \omega_{syn} \sum_{i=1}^{n} (d_i + r_{t,i}^{-1} - \hat{c}_i(0))$. Plugging these two equations to the definition of ES in (9) yields (23).

2) Power Fluctuations and Measurement Noise: We seek to characterize the effect of power fluctuations and frequency measurement noise on the frequency variance, i.e., the \mathcal{H}_2 norm of the system $\hat{T}_{\omega dn}$.

We first show that the squared \mathcal{H}_2 norm of $\hat{T}_{\omega dn}$ is a weighted sum of the squared \mathcal{H}_2 norm of each $\hat{h}_{p,k}$ and $\hat{h}_{\omega,k}$ in the diagonalized system (21).

Theorem 2. (Frequency variance): Define $\Gamma := V^T F^{-1} V$. If Assumptions 1 and 2 hold, then

$$\|\hat{T}_{\omega dn}\|_{\mathcal{H}_{2}}^{2} = \sum_{k=1}^{n} \Gamma_{kk} \left(\kappa_{p}^{2} \|\hat{h}_{p,k}\|_{\mathcal{H}_{2}}^{2} + \kappa_{\omega}^{2} \|\hat{h}_{\omega,k}\|_{\mathcal{H}_{2}}^{2} \right).$$

Proof: It follows from (8) and (10) that

$$\begin{split} \|\hat{T}_{\omega dn}\|_{\mathcal{H}_{2}}^{2} &= \frac{1}{2\pi} \int_{-\infty}^{\infty} \operatorname{tr} \left(\hat{T}_{\omega d}(j\omega)^{*} \hat{T}_{\omega d}(j\omega) \right) \, \mathrm{d}\omega \\ &+ \frac{1}{2\pi} \int_{-\infty}^{\infty} \operatorname{tr} \left(\hat{T}_{\omega n}(j\omega)^{*} \hat{T}_{\omega n}(j\omega) \right) \, \mathrm{d}\omega \\ &=: \|\hat{T}_{\omega d}\|_{\mathcal{H}_{2}}^{2} + \|\hat{T}_{\omega n}\|_{\mathcal{H}_{2}}^{2}. \end{split}$$

We now compute $\|\hat{T}_{\omega d}\|_{\mathcal{H}_2}^2$. Using (21b) and the fact that $\hat{W}_{\mathrm{p}}(s) = \kappa_{\mathrm{p}} F^{\frac{1}{2}}$ by Assumption 2, we get $\hat{T}_{\omega \mathrm{d}}(s) = \kappa_{\mathrm{p}} F^{-\frac{1}{2}} V \hat{H}_{\mathrm{p}}(s) V^T$. Therefore,

 $\hat{T}_{\omega d}(j\omega)^* \hat{T}_{\omega d}(j\omega) = \kappa_p^2 V \hat{H}_p(j\omega)^* V^T F^{-1} V \hat{H}_p(j\omega) V^T.$ Using the cyclic property of the trace, this implies that

$$\operatorname{tr}\left(\hat{T}_{\omega \mathrm{d}}(j\omega)^{*}\hat{T}_{\omega \mathrm{d}}(j\omega)\right) = \kappa_{\mathrm{p}}^{2} \operatorname{tr}\left(\hat{H}_{\mathrm{p}}(j\omega)^{*}\Gamma\hat{H}_{\mathrm{p}}(j\omega)\right)$$

where $\Gamma := V^T F^{-1} V$. Therefore, it follows that

=

$$\begin{split} \|\hat{T}_{\omega \mathrm{d}}\|_{\mathcal{H}_{2}}^{2} &= \frac{1}{2\pi} \int_{-\infty}^{\infty} \kappa_{\mathrm{p}}^{2} \mathrm{tr} \left(\hat{H}_{\mathrm{p}}(j\omega)^{*} \Gamma \hat{H}_{\mathrm{p}}(j\omega) \right) \,\mathrm{d}\omega \\ &= \sum_{k=1}^{n} \frac{\kappa_{\mathrm{p}}^{2} \Gamma_{kk}}{2\pi} \int_{-\infty}^{\infty} \left| \hat{h}_{\mathrm{p},k}(j\omega) \right|^{2} \,\mathrm{d}\omega = \kappa_{\mathrm{p}}^{2} \sum_{k=1}^{n} \Gamma_{kk} \|\hat{h}_{\mathrm{p},k}\|_{\mathcal{H}_{2}}^{2}. \end{split}$$

The result follows from a similar argument on $\|\hat{T}_{\omega n}\|_{\mathcal{H}_2}^2$.

Theorem 2 allows us to compute the \mathcal{H}_2 norm of $T_{\omega dn}$ by means of computing the norms of a set of simple scalar transfer functions. However, for different controllers, the transfer functions $\hat{h}_{p,k}$ and $\hat{h}_{\omega,k}$ will change. Since in all the cases these transfer functions are of fourth-order or lower, the following lemma will suffice for the purpose of our comparison.

Lemma 2. (\mathcal{H}_2 norm of a fourth-order transfer function): Let

$$\hat{h}(s) = \frac{b_3 s^3 + b_2 s^2 + b_1 s + b_0}{s^4 + a_3 s^3 + a_2 s^2 + a_1 s + a_0} + b_4$$

be a stable transfer function. If $b_4 = 0$, then

$$\|\hat{h}\|_{\mathcal{H}_2}^2 = \frac{\zeta_0 b_0^2 + \zeta_1 b_1^2 + \zeta_2 b_2^2 + \zeta_3 b_3^2 + \zeta_4}{2a_0 \left(a_1 a_2 a_3 - a_1^2 - a_0 a_3^2\right)}$$
(24)

where

$$\begin{aligned} \zeta_0 &:= a_2 a_3 - a_1, \quad \zeta_1 &:= a_0 a_3, \quad \zeta_2 &:= a_0 a_1, \\ \zeta_3 &:= a_0 a_1 a_2 - a_0^2 a_3, \quad \zeta_4 &:= -2a_0 (a_1 b_1 b_3 + a_3 b_0 b_2). \end{aligned}$$
(25)

Otherwise, $\|\hat{h}\|_{\mathcal{H}_2}^2 = \infty$.

Proof: See Appendix B.

Remark 6. (\mathcal{H}_2 norm of a transfer function lower than fourthorder): Although Lemma 2 is stated for a fourth-order transfer function, it can also be used to find the \mathcal{H}_2 norm of third-, second-, and first-order transfer functions by considering appropriate limits. For example, setting $a_0 = b_0 = \epsilon$ and considering the limit $\epsilon \to 0$, (24) gives the \mathcal{H}_2 norm of a generic third-order transfer function. This process shows that given a stable transfer function $\hat{h}(s)$, if $b_4 = 0$ and

• (third-order) $a_0 = b_0 = 0$, then

$$\|\hat{h}\|_{\mathcal{H}_2}^2 = \frac{a_3b_1^2 + a_1b_2^2 + a_1a_2b_3^2 - 2a_1b_1b_3}{2a_1(a_2a_3 - a_1)}$$

• (second-order) $a_0 = b_0 = a_1 = b_1 = 0$, then $b_0^2 = a_0 b_1^2 = b_0^2 b_1^2$

$$\|\hat{h}\|_{\mathcal{H}_2}^2 = \frac{b_2^2 + a_2 b_3^2}{2a_2 a_3}$$

• (first-order) $a_0 = b_0 = a_1 = b_1 = a_2 = b_2 = 0$, then

$$\|\hat{h}\|_{\mathcal{H}_2}^2 = rac{b_3^2}{2a_3}$$

otherwise $\|\hat{h}\|_{\mathcal{H}_2}^2 = \infty$.

Remark 7. (Well-definedness by the stability): Note that the stability of $\hat{h}(s)$ guarantees that the denominators in all the above \mathcal{H}_2 norm expressions are nonzero by the Routh-Hurwitz stability criterion.

3) Synchronization Cost: The computation of the synchronization cost defined in (11) for the system $\hat{T}_{\omega p}$ in the absence of inverter control can be found in [13]. Taking this into account, we can get corresponding results for the system with any control law readily.

Lemma 3. (Synchronization cost): Let Assumptions 1 and 3 hold. Define $\tilde{u}_0 := V_{\perp}^T F^{-\frac{1}{2}} u_0$ and $\tilde{\Gamma} := V_{\perp}^T F^{-1} V_{\perp}$. Then, the synchronization cost of the system $\hat{T}_{\omega p}$ is given by

$$\|\tilde{\omega}\|_2^2 = \tilde{u}_0^T \left(\tilde{\Gamma} \circ \tilde{H}\right) \tilde{u}_0$$

where \circ denotes the Hadamard product and $\tilde{H} \in \mathbb{R}^{(n-1)\times(n-1)}$ is the matrix with entries

$$\tilde{H}_{kl} := \int_0^\infty h_{\mathbf{u},k}(t) h_{\mathbf{u},l}(t) \,\mathrm{d}t \,, \quad \forall k,l \in \{1,\ldots,n-1\}$$

with $\hat{h}_{u,k}(s) := \hat{h}_{p,k+1,T}(s)/s$ and $\hat{h}_{p,k,T}(s)$ being a specified case of the transfer function $\hat{h}_{p,k}(s)$ defined in (19), i.e., when the turbine is triggered.

Proof: This is a direct extension of [13, Prop. 2].

Lemma 3 shows that the computation of the synchronization cost requires knowing the inner products \tilde{H}_{kl} . Yet, the general expressions of these inner products for an arbitrary combination of k and l are too tedious to be useful in analysis. Thus, we will investigate instead bounds on the synchronization cost in terms of the inner products \tilde{H}_{kl} when k = l, which are exactly the \mathcal{H}_2 norms of transfer functions $\hat{h}_{u,k}(s)$.

Lemma 4. (Bounds for Hadamard product): Let $P \in \mathbb{R}^{n \times n}$ be a symmetric matrix with minimum and maximum eigenvalues given by $\lambda_{\min}(P)$ and $\lambda_{\max}(P)$, respectively. Then, $\forall x, y \in \mathbb{R}^n$

$$\lambda_{\min}(P) \sum_{k=1}^{n} x_k^2 y_k^2 \le x^T \left(P \circ \left(y y^T \right) \right) x \le \lambda_{\max}(P) \sum_{k=1}^{n} x_k^2 y_k^2.$$
Proof: See Appendix C.

Lemma 4 implies the bounds on the synchronization cost.

Theorem 3. (Bounds on synchronization cost): Let Assumptions 1 and 3 hold. Then, the synchronization cost of the system $\hat{T}_{\omega p}$ is bounded by $\|\tilde{\omega}\|_2^2 \leq \|\tilde{\omega}\|_2^2 \leq \|\tilde{\omega}\|_2^2$, where

$$\begin{split} & \underline{\|\tilde{\omega}\|_{2}^{2}} := \frac{\sum_{k=1}^{n-1} \tilde{u}_{0,k}^{2} \|\hat{h}_{\mathbf{u},k}\|_{\mathcal{H}_{2}}^{2}}{\max_{i \in \mathcal{V}} (f_{i})} \text{ and } \\ & \overline{\|\tilde{\omega}\|_{2}^{2}} := \frac{\sum_{k=1}^{n-1} \tilde{u}_{0,k}^{2} \|\hat{h}_{\mathbf{u},k}\|_{\mathcal{H}_{2}}^{2}}{\min_{i \in \mathcal{V}} (f_{i})}. \end{split}$$

Proof: By Lemma 3,

$$\begin{split} \|\tilde{\omega}\|_{2}^{2} &= \int_{0}^{\infty} \tilde{u}_{0}^{T} \left(\tilde{\Gamma} \circ \left(h_{\mathrm{u}}(t)h_{\mathrm{u}}(t)^{T}\right)\right) \tilde{u}_{0} \,\mathrm{d}t \\ &\geq \int_{0}^{\infty} \lambda_{\min}(\tilde{\Gamma}) \sum_{k=1}^{n-1} \tilde{u}_{0,k}^{2} h_{\mathrm{u},k}(t)^{2} \,\mathrm{d}t \\ &= \lambda_{\min}(\tilde{\Gamma}) \sum_{k=1}^{n-1} \tilde{u}_{0,k}^{2} \|\hat{h}_{\mathrm{u},k}\|_{\mathcal{H}_{2}}^{2} \\ &\geq \lambda_{\min}(F^{-1}) \sum_{k=1}^{n-1} \tilde{u}_{0,k}^{2} \|\hat{h}_{\mathrm{u},k}\|_{\mathcal{H}_{2}}^{2} = \frac{\sum_{k=1}^{n-1} \tilde{u}_{0,k}^{2} \|\hat{h}_{\mathrm{u},k}\|_{\mathcal{H}_{2}}^{2}}{\max_{i \in \mathcal{V}} (f_{i})} \end{split}$$

which concludes the proof of the lower bound. The first inequality follows from Lemma 4 by setting $P = \tilde{\Gamma}$, $x = \tilde{u}_0$, and $y = h_u(t) := (h_{u,k}(t), k \in \{1, \dots, n-1\}) \in \mathbb{R}^{n-1}$. The second inequality follows from the interlacing theorem [31, Th. 4.3.17]. The proof of the upper bound is similar.

Remark 8. (Synchronization cost in homogeneous case): In the system with homogeneous parameters, i.e., $F = fI_n$ for some f > 0, the identical lower and upper bounds on the synchronization cost imply that

$$\|\tilde{\omega}\|_2^2 = f^{-1} \sum_{k=1}^{n-1} \tilde{u}_{0,k}^2 \|\hat{h}_{\mathbf{u},k}\|_{\mathcal{H}_2}^2.$$

4) Nadir: A deep Nadir poses a threat to the reliable operation of a power system. Hence, one of the goals of inverter control laws is the reduction of Nadir. We seek to evaluate the ability of different control laws to eliminate Nadir. To this end, we provide a necessary and sufficient condition for Nadir elimination in a second-order system with a zero.

Theorem 4. (Nadir elimination for a second-order system): Assume K > 0, z > 0, $\xi \ge 0$, $\omega_n > 0$. The step response of a second-order system with transfer function given by

$$\hat{h}(s) = \frac{K(s+z)}{s^2 + 2\xi\omega_{\rm n}s + \omega_{\rm n}^2}$$

has no Nadir if and only if

$$1 \le \xi \le z/\omega_{\rm n} \quad \text{or} \quad \begin{cases} \xi > z/\omega_{\rm n} \\ \xi \ge (z/\omega_{\rm n} + \omega_{\rm n}/z)/2 \end{cases}$$
(26)

where the conditions in braces jointly imply $\xi > 1$.

Proof: Basically, Nadir must occur at some non-negative finite time instant t_{nadir} , such that $\dot{p}_u(t_{\text{nadir}}) = 0$ and $p_u(t_{\text{nadir}})$ is a maximum, where $p_u(t)$ denotes the unit-step response of $\hat{h}(s)$, i.e., $\hat{p}_u(s) := \hat{h}(s)/s$. We consider three cases based on the value of damping ratio ξ separately.

1) Under damped case $(0 \le \xi < 1)$: The output is

$$\hat{p}_{\mathbf{u}}(s) = \frac{Kz}{\omega_{\mathbf{n}}^2} \left[\frac{1}{s} - \frac{s + \xi\omega_{\mathbf{n}}}{(s + \xi\omega_{\mathbf{n}})^2 + \omega_{\mathbf{d}}^2} - \frac{\xi\omega_{\mathbf{n}} - \omega_{\mathbf{n}}^2 z^{-1}}{(s + \xi\omega_{\mathbf{n}})^2 + \omega_{\mathbf{d}}^2} \right]$$

with $\omega_d := \omega_n \sqrt{1 - \xi^2}$, which gives the time domain response

$$p_{\rm u}(t) = \frac{Kz}{\omega_{\rm n}^2} \left[1 - e^{-\xi \omega_{\rm n} t} \eta_0 \sin\left(\omega_{\rm d} t + \phi\right) \right]$$

where

$$\eta_0 = \sqrt{1 + \frac{\left(\xi\omega_n - \omega_n^2 z^{-1}\right)^2}{\omega_d^2}} \text{ and } \tan \phi = \frac{\omega_d}{\xi\omega_n - \omega_n^2 z^{-1}}.$$

Clearly, the above response must have oscillations. Therefore, for the case $0 \le \xi < 1$, Nadir always exists.

2) Critically damped case ($\xi = 1$): The output is

$$\hat{p}_{\mathbf{u}}(s) = \frac{Kz}{\omega_{\mathbf{n}}^2} \left[\frac{1}{s} - \frac{1}{s + \omega_{\mathbf{n}}} - \frac{\omega_{\mathbf{n}} - \omega_{\mathbf{n}}^2 z^{-1}}{(s + \omega_{\mathbf{n}})^2} \right]$$

which gives the time domain response

$$p_{\mathbf{u}}(t) = \frac{Kz}{\omega_{\mathbf{n}}^2} \left\{ 1 - e^{-\omega_{\mathbf{n}}t} \left[1 + \left(\omega_{\mathbf{n}} - \omega_{\mathbf{n}}^2 z^{-1}\right) t \right] \right\} .$$

Thus,

$$\dot{p}_{\mathrm{u}}(t) = K z e^{-\omega_{\mathrm{n}} t} \left[\left(1 - \omega_{\mathrm{n}} z^{-1} \right) t + z^{-1} \right] \,.$$

Letting $\dot{p}_{u}(t) = 0$ yields

 $\omega_{n}e^{-\omega_{n}t}\left[1+\left(\omega_{n}-\omega_{n}^{2}z^{-1}\right)t\right]=e^{-\omega_{n}t}\left(\omega_{n}-\omega_{n}^{2}z^{-1}\right)$ which has a non-negative finite solution

$$t_{\rm nadir} = \frac{z^{-1}}{\omega_{\rm n} z^{-1} - 1}$$

whenever $\omega_n z^{-1} > 1$. For any $\epsilon > 0$, it holds that $\dot{p}_u(t_{\text{nadir}} - \epsilon) = \epsilon K z e^{-\omega_n(t_{\text{nadir}} - \epsilon)} (\omega_n z^{-1} - 1) > 0$

 $\dot{p}_{\rm u}(t_{\rm nadir}+\epsilon) = \epsilon K z e^{-\omega_{\rm n}(t_{\rm nadir}+\epsilon)} \left(1-\omega_{\rm n} z^{-1}\right) < 0 \; . \\ \mbox{Clearly, Nadir occurs at } t_{\rm nadir}. \mbox{ Therefore, for the case} \\ \xi = 1, \mbox{ Nadir is eliminated if and only if } \omega_{\rm n} z^{-1} \leq 1. \mbox{ To put it more succinctly, we combine the two conditions} into$

$$1 = \xi \le z/\omega_{\rm n}.\tag{27}$$

3) Over damped case ($\xi > 1$): The output is

$$\hat{p}_{\mathrm{u}}(s) = \frac{Kz}{\omega_{\mathrm{n}}^2} \left(\frac{1}{s} - \frac{\eta_1}{s + \sigma_1} - \frac{\eta_2}{s + \sigma_2} \right)$$

with

$$\sigma_{1,2} = \omega_n \left(\xi \pm \sqrt{\xi^2 - 1} \right) \text{ and } \eta_{1,2} = \frac{1}{2} \mp \frac{\xi - \omega_n z^{-1}}{2\sqrt{\xi^2 - 1}}$$

which gives the time domain response

$$p_{\rm u}(t) = \frac{Kz}{\omega_{\rm n}^2} \left(1 - \eta_1 e^{-\sigma_1 t} - \eta_2 e^{-\sigma_2 t} \right).$$

Thus,

$$\dot{p}_{\mathrm{u}}(t) = \frac{Kz}{\omega_{\mathrm{n}}^2} \left(\sigma_1 \eta_1 e^{-\sigma_1 t} + \sigma_2 \eta_2 e^{-\sigma_2 t} \right).$$

Letting $\dot{p}_{\rm u}(t) = 0$ yields $\sigma_1 \eta_1 e^{-\sigma_1 t} = -\sigma_2 \eta_2 e^{-\sigma_2 t}$, which has a non-negative finite solution

$$t_{\text{nadir}} = \frac{1}{2\omega_{\text{n}}\sqrt{\xi^2 - 1}} \ln \frac{1 - \omega_{\text{n}} z^{-1} \left(\xi + \sqrt{\xi^2 - 1}\right)}{1 - \omega_{\text{n}} z^{-1} \left(\xi - \sqrt{\xi^2 - 1}\right)}$$

whenever $1 - \omega_n z^{-1} (\xi - \sqrt{\xi^2 - 1}) < 0$. For any $\epsilon > 0$, it holds that

$$\begin{split} \dot{p}_{\mathrm{u}}(t_{\mathrm{nadir}} - \epsilon) &> \frac{Kz}{\omega_{\mathrm{n}}^{2}} e^{\sigma_{1}\epsilon} \left(\sigma_{1}\eta_{1}e^{-\sigma_{1}t_{\mathrm{nadir}}}\right. \\ &+ \sigma_{2}\eta_{2}e^{-\sigma_{2}t_{\mathrm{nadir}}}\right) \\ &= e^{\sigma_{1}\epsilon}\dot{p}_{\mathrm{u}}(t_{\mathrm{nadir}}) = 0 \\ \dot{p}_{\mathrm{u}}(t_{\mathrm{nadir}} + \epsilon) &< \frac{Kz}{\omega_{\mathrm{n}}^{2}}e^{-\sigma_{1}\epsilon} \left(\sigma_{1}\eta_{1}e^{-\sigma_{1}t_{\mathrm{nadir}}} \right. \\ &+ \sigma_{2}\eta_{2}e^{-\sigma_{2}t_{\mathrm{nadir}}}\right) \\ &= e^{-\sigma_{1}\epsilon}\dot{p}_{\mathrm{u}}(t_{\mathrm{nadir}}) = 0 \end{split}$$

since $\sigma_1 > \sigma_2 > 0$ and one can show that $\sigma_2 \eta_2 < 0$. Clearly, Nadir occurs at t_{nadir} . Therefore, for the case $\xi > 1$, Nadir is eliminated if and only if $1 - \omega_n z^{-1} (\xi - \sqrt{\xi^2 - 1}) \ge 0$, i.e., $\sqrt{\xi^2 - 1} \ge \xi - z/\omega_n$, which holds if and only if

$$\xi \leq z/\omega_{\rm n}$$
 or $\begin{cases} \xi > z/\omega_{\rm n} \\ \xi \geq (z/\omega_{\rm n} + \omega_{\rm n}/z)/2. \end{cases}$

Thus, we get the conditions

$$1 < \xi \le z/\omega_{n} \quad \text{or} \quad \begin{cases} \xi > 1 \\ \xi > z/\omega_{n} \\ \xi \ge (z/\omega_{n} + \omega_{n}/z)/2. \end{cases}$$
(28)

Finally, since $\forall a, b \ge 0$, $(a + b)/2 \ge \sqrt{ab}$ with equality only when a = b, it follows that the second condition in (28) can only hold when $\xi > 1$. Thus, we can combine (27) and (28) to yield (26).

IV. THE NEED FOR A BETTER SOLUTION

We now apply the results in Section III to illustrate the performance limitations of the traditional control laws DC and VI. With this aim, we seek to quantify the frequency variance (10) under DC and VI through the \mathcal{H}_2 norm of $\hat{T}_{\omega dn,DC}$ and $\hat{T}_{\omega dn,VI}$, as well as the steady-state effort share (9), synchronization cost (11), and Nadir (12) through the step response characterizations of $\hat{T}_{\omega p,DC}$ and $\hat{T}_{\omega p,VI}$.

A. Steady-State Effort Share

Corollary 1. (Synchronous frequency under DC and VI): Let Assumption 3 hold. When $q_{r,i}$ is defined by the control law DC (5) or VI (6), the steady-state frequency deviation of the system $\hat{T}_{\omega p, \text{DC}}$ or $\hat{T}_{\omega p, \text{VI}}$ synchronizes to the synchronous frequency, i.e., $\omega_{ss} = \omega_{syn} \mathbb{1}_n$ with

$$\omega_{\rm syn} = \frac{\sum_{i=1}^{n} u_{0,i}}{\sum_{i=1}^{n} \left(d_i + r_{\rm t,i}^{-1} + r_{\rm r,i}^{-1} \right)} \,. \tag{29}$$

Proof: The result follows directly from Lemma 1.

Now, the corollary below gives the expression for the steadystate effort share when inverters are under the control law DC or VI.

Corollary 2. (Steady-state effort share of DC and VI): Let Assumption 3 hold. If $q_{r,i}$ is under the control law (5) or (6), then the steady-state effort share of the system $\hat{T}_{\omega p,DC}$ or $\hat{T}_{\omega p,VI}$ is given by

$$ES = \frac{\sum_{i=1}^{n} r_{r,i}^{-1}}{\sum_{i=1}^{n} \left(d_i + r_{t,i}^{-1} + r_{r,i}^{-1} \right)}.$$
(30)

Proof: The result follows directly from Theorem 1 applied to (5) and (6).

Corollary 2 indicates that DC and VI have the same steadystate effort share, which increases as $r_{r,i}^{-1}$ increase. However, $r_{r,i}^{-1}$ are parameters that also directly affect the dynamic performance of the power system, which can be seen clearly from the dynamic performance analysis.

B. Power Fluctuations and Measurement Noise

Using Theorem 2 and Lemma 2, it is possible to get closedform expressions of \mathcal{H}_2 norms for systems $\hat{T}_{\omega dn,DC}$ and $\hat{T}_{\omega dn,VI}$.

Corollary 3. (Frequency variance under DC and VI): Let Assumptions 1 and 2 hold. The squared \mathcal{H}_2 norm of $\hat{T}_{\omega dn,DC}$ and $\hat{T}_{\omega dn,VI}$ is given by

$$\|\hat{T}_{\omega dn, DC}\|_{\mathcal{H}_{2}}^{2} = \sum_{k=1}^{n} \Gamma_{kk} \frac{\kappa_{p}^{2} + r_{r}^{-2} \kappa_{\omega}^{2}}{2m\check{d}}$$
(31a)

$$\|\hat{T}_{\omega dn, VI}\|_{\mathcal{H}_2}^2 = \infty \tag{31b}$$

respectively, where $\check{d} := d + r_r^{-1}$.

Proof: We study the two cases separately.

We begin with $\|\hat{T}_{\omega dn,DC}\|_{\mathcal{H}_2}^2$. Applying (13) and (15) to (19) and (20) shows $\hat{h}_{p,k,DC}(s)$ is a transfer function with $b_4 = a_0 = b_0 = a_1 = b_1 = 0, a_2 = \lambda_k/m, b_2 = 0, a_3 = \check{d}/m, b_3 = 1/m$, while $\hat{h}_{\omega,k,DC}(s)$ is a transfer function with $b_4 = a_0 = b_0 = a_1 = b_1 = 0, a_2 = \lambda_k/m, b_2 = 0, a_3 = \check{d}/m, b_3 = -r_r^{-1}/m$. Thus, by Lemma 2

$$\|\hat{h}_{\mathrm{p},k,\mathrm{DC}}\|_{\mathcal{H}_2}^2 = \frac{1}{2m\check{d}} \text{ and } \|\hat{h}_{\omega,k,\mathrm{DC}}\|_{\mathcal{H}_2}^2 = \frac{r_{\mathrm{r}}^{-2}}{2m\check{d}}$$

Then, (31a) follows from Theorem 2.

We now turn to show that $\|\hat{T}_{\omega dn, VI}\|_{\mathcal{H}_2}^2$ is infinite. Define $\check{m} := m + m_v$. Applying (13) and (16) to (20) yields

$$\hat{h}_{\omega,k, ext{VI}}(s) = -rac{m_{ ext{v}}s^2 + r_{ ext{r}}^{-1}s}{\check{m}s^2 + \check{d}s + \lambda_k}$$

which by Lemma 2 has $b_4 = -m_v/\check{m} \neq 0$ and thus $\|\hat{h}_{\omega,k,\text{DC}}\|_{\mathcal{H}_2}^2 = \infty$. Then, (31b) follows from Theorem 2.

Corollary 4. (Optimal r_r^{-1} for $\|\hat{T}_{\omega dn,DC}\|_{\mathcal{H}_2}^2$): Let Assumptions 1 and 2 hold. Then

$$r_{\rm r}^{-1\star} := \operatorname{argmin}_{r_{\rm r}^{-1} > 0} \| \hat{T}_{\omega dn, {\rm DC}} \|_{\mathcal{H}_2}^2 = -d + \sqrt{d^2 + (\kappa_{\rm p}/\kappa_{\omega})^2}$$
(32)

$$\partial_{r_{\rm r}^{-1}} \| \hat{T}_{\omega {\rm dn,DC}} \|_{\mathcal{H}_2}^2 = \sum_{k=1}^n \Gamma_{kk} \frac{\kappa_{\omega}^2 r_{\rm r}^{-2} + 2d\kappa_{\omega}^2 r_{\rm r}^{-1} - \kappa_{\rm p}^2}{2m\check{d}^2} \,. \tag{33}$$

By equating (33) to 0, we can solve the corresponding r_r^{-1} as $r_r^{-1*}{}_{\pm} = -d \pm \sqrt{d^2 + (\kappa_p/\kappa_\omega)^2}$. The only positive root is therefore $r_r^{-1*} := -d + \sqrt{d^2 + (\kappa_p/\kappa_\omega)^2}$. We now show that $\Gamma_{kk} > 0, \forall k \in \{1, \ldots, n\}$. Recall that $\Gamma := V^T F^{-1} V$. We know $\Gamma_{kk} = \sum_{j=1}^n (v_{k,j}^2/f_j)$. Since v_k is an eigenvector, $\forall k \in \{1, \ldots, n\}$, there must exist at least one $j \in \mathcal{V}$ such that $v_{k,j} \neq 0$. Since $f_i > 0, \forall i$, we have that $\Gamma_{kk} > 0, \forall k \in \{1, \ldots, n\}$. In addition, since the denominator of (33) is always positive and the highest order coefficient of the numerator is positive, whenever $0 < r_r^{-1} < r_r^{-1*}$, then $\partial_{r_r^{-1}} \|\hat{T}_{\omega dn, DC}\|_{\mathcal{H}_2}^2 < 0$, and if $r_r^{-1} > r_r^{-1*}$, then $\partial_{r_r^{-1}} \|\hat{T}_{\omega dn, DC}\|_{\mathcal{H}_2}^2 > 0$. Therefore, r_r^{-1*} is the minimizer of $\|\hat{T}_{\omega dn, DC}\|_{\mathcal{H}_2}^2$.

Two main observations can be made from Corollary 3. First, the control parameter r_r^{-1} of DC has an direct effect on the size of the frequency variance in the system, which makes it impossible to require DC to bear an assigned amount of steadystate effort share and reduce the frequency variance at the same time. The other important point is that VI will induce unbounded frequency variance, which poses a threat to the operation of the power system. Therefore, neither DC nor VI is a good solution to improve the frequency variance without sacrificing the steadystate effort share.

C. Synchronization Cost

Theorem 3 implies that the synchronization cost of $\hat{T}_{\omega \mathrm{p,DC}}$ and $\hat{T}_{\omega \mathrm{p,VI}}$ is bounded by a weighted sum of $\|\hat{h}_{\mathrm{u},k,\mathrm{DC}}\|_{\mathcal{H}_2}^2$ and $\|\hat{h}_{\mathrm{u},k,\mathrm{VI}}\|_{\mathcal{H}_2}^2$, respectively. Hence, in order to see the limited ability of DC and VI to reduce the synchronization cost, we need to gain a deeper understanding of $\|\hat{h}_{\mathrm{u},k,\mathrm{DC}}\|_{\mathcal{H}_2}^2$ and $\|\hat{h}_{\mathrm{u},k,\mathrm{VI}}\|_{\mathcal{H}_2}^2$ first.

Theorem 5. (Bounds of $\|\hat{h}_{u,k,DC}\|_{\mathcal{H}_2}^2$ and $\|\hat{h}_{u,k,VI}\|_{\mathcal{H}_2}^2$): Let Assumptions 1 and 3 hold. Then, given $r_r^{-1} > 0, \forall m_v > 0$,

$$\frac{1}{2\lambda_{k+1} \left(\check{d} + r_{t}^{-1} \right)} < \| \hat{h}_{u,k,VI} \|_{\mathcal{H}_{2}}^{2} < \| \hat{h}_{u,k,DC} \|_{\mathcal{H}_{2}}^{2} < \| \hat{h}_{u,k,SW} \|_{\mathcal{H}_{2}}^{2}$$

where $\|\hat{h}_{u,k,SW}\|_{\mathcal{H}_2}^2$ represents the inner products of the open-loop system with no additional control from inverters.

Proof: Considering that DC can be viewed as VI with $m_v = 0$ and the open-loop system can be viewed as VI with $m_v = r_r^{-1} = 0$, we only compute $\|\hat{h}_{u,k,VI}\|_{\mathcal{H}_2}^2$, which straightforwardly implies the other two. Applying (14) and (16) to (19) shows $\hat{h}_{u,k,VI}(s) = \hat{h}_{p,k+1,T,VI}(s)/s$ is a transfer function with $b_4 = a_0 = b_0 = 0, a_1 = \lambda_{k+1}/(\check{m}\tau), b_1 = 1/(\check{m}\tau), a_2 = (\check{d} + r_t^{-1} + \lambda_{k+1}\tau)/(\check{m}\tau), b_2 = 1/\check{m}, a_3 = (\check{m} + \check{d}\tau)/(\check{m}\tau), b_3 = 0$. Then, it follows from Lemma 2 that

$$\|\hat{h}_{\mathrm{u},k,\mathrm{VI}}\|_{\mathcal{H}_{2}}^{2} = \frac{\check{m} + \tau \left(\lambda_{k+1}\tau + \check{d}\right)}{2\lambda_{k+1} \left[\tau \check{d} \left(\lambda_{k+1}\tau + \check{d} + r_{\mathrm{t}}^{-1}\right) + \check{m} \left(\check{d} + r_{\mathrm{t}}^{-1}\right)\right]}$$

Since $\|\hat{h}_{u,k,VI}\|_{\mathcal{H}_2}^2$ is a function of r_r^{-1} and m_v , in what follows we denote it by $\rho(r_r^{-1}, m_v)$. In order to have an insight on how $\|\hat{h}_{u,k,VI}\|_{\mathcal{H}_2}^2$ changes with r_r^{-1} and m_v , we take partial derivatives of $\rho(r_r^{-1}, m_v)$ with respect to r_r^{-1} and m_v , i.e.,

$$\begin{aligned} \partial_{r_{\mathrm{r}}^{-1}}\rho(r_{\mathrm{r}}^{-1},m_{\mathrm{v}}) \\ &= -\frac{\left[\check{m}+\tau\left(\lambda_{k+1}\tau+\check{d}\right)\right]^{2}+\lambda_{k+1}\tau^{3}r_{\mathrm{t}}^{-1}}{2\lambda_{k+1}\left[\tau\check{d}\left(\lambda_{k+1}\tau+\check{d}+r_{\mathrm{t}}^{-1}\right)+\check{m}(\check{d}+r_{\mathrm{t}}^{-1})\right]^{2}} \\ \partial_{m_{\mathrm{v}}}\rho(r_{\mathrm{r}}^{-1},m_{\mathrm{v}}) \\ &= -\frac{\tau^{2}r_{\mathrm{t}}^{-1}}{2\left[\tau\check{d}\left(\lambda_{k+1}\tau+\check{d}+r_{\mathrm{t}}^{-1}\right)+\check{m}(\check{d}+r_{\mathrm{t}}^{-1})\right]^{2}}.\end{aligned}$$

Clearly, for all $r_r^{-1} \ge 0$, $\partial_{r_r^{-1}}\rho(r_r^{-1}, m_v) < 0$, which means that $\rho(r_r^{-1}, m_v)$ is a monotonically decreasing function of r_r^{-1} . Similarly, for all $m_v \ge 0$, $\partial_{m_v}\rho(r_r^{-1}, m_v) < 0$, which means that $\rho(r_r^{-1}, m_v)$ is a monotonically decreasing function of m_v . Therefore, given $r_r^{-1} > 0$, $\forall m_v > 0$, it holds that

$$\lim_{m_{\mathbf{v}}\to\infty}\rho(r_{\mathbf{r}}^{-1},m_{\mathbf{v}}) < \rho(r_{\mathbf{r}}^{-1},m_{\mathbf{v}}) < \rho(r_{\mathbf{r}}^{-1},0) < \rho(0,0).$$

Recall that $\|\hat{h}_{u,k,VI}\|_{\mathcal{H}_2}^2 = \rho(r_r^{-1}, m_v), \|\hat{h}_{u,k,DC}\|_{\mathcal{H}_2}^2 = \rho(r_r^{-1}, 0), \text{ and } \|\hat{h}_{u,k,SW}\|_{\mathcal{H}_2}^2 = \rho(0, 0).$ The result follows.

Corollary 5. (Comparison of synchronization cost in homogeneous case): Denote the synchronization cost of the open-loop system as $\|\tilde{\omega}_{\rm SW}\|_2^2$. Then, under Assumptions 1 and 3, given $r_r^{-1} > 0, \forall m_v > 0$, we can order the synchronization cost when $F = fI_n$ as

$$\frac{\sum_{k=1}^{n-1} \left(\tilde{u}_{0,k}^2 / \lambda_{k+1} \right)}{2f \left(\check{d} + r_{\mathrm{t}}^{-1} \right)} < \| \tilde{\omega}_{\mathrm{VI}} \|_2^2 < \| \tilde{\omega}_{\mathrm{DC}} \|_2^2 < \| \tilde{\omega}_{\mathrm{SW}} \|_2^2.$$

Proof: The result follows by combining Remark 8 and Theorem 5.

Corollary 6. (Lower bound of synchronization cost under DC and VI): Under Assumptions 1 and 3, the ordering of the size of the bounds on the synchronization cost of open-loop, DC, and VI depends on the parameter values. Thus, we cannot order $\|\tilde{\omega}_{\rm VI}\|_2^2$, $\|\tilde{\omega}_{\rm DC}\|_2^2$, and $\|\tilde{\omega}_{\rm SW}\|_2^2$ strictly. Instead, we highlight that, given $r_{\rm r}^{-1} > 0$, the synchronization cost under DC and VI is bounded below by

$$\frac{\sum_{k=1}^{n-1} \left(\tilde{u}_{0,k}^2 / \lambda_{k+1}\right)}{2 \max_{i \in \mathcal{V}} \left(f_i\right) \left(\check{d} + r_{\mathrm{t}}^{-1}\right)} \,.$$

Proof: The result follows from Theorems 3 and 5.

Corollary 5 provides both upper and lower bounds for the synchronization cost under DC and VI in homogeneous case. The upper bound verifies that DC and VI do reduce the synchronization cost by adding damping and inertia while the lower bound indicates that the reduction of the synchronization cost through DC and VI is limited by certain value that is dependent on r_r^{-1} . Corollary 6 implies that in the proportional case, the synchronization cost under DC and VI is also bounded below by a value that is dependent on r_r^{-1} . The fact that the lower bound of the synchronization cost under DC and VI is reduced as r_r^{-1} increases is unsatisfactory. This is because increasing r_r^{-1} also increases the steady-state effort share, which is usually not

desired. However, given a small r_r^{-1} , even if the inertia is very high, i.e., $m_v \to \infty$, the synchronization cost $\|\tilde{\omega}_{VI}\|_2^2$ can never reach zero, not to mention $\|\tilde{\omega}_{DC}\|_2^2$.

D. Nadir

Finally, with the help of Theorem 4, we can determine the conditions that the parameters of DC and VI must satisfy to eliminate Nadir of the system frequency.

Theorem 6. (Nadir elimination under DC and VI): Under Assumptions 1 and 3:

• for $\hat{T}_{\omega p,DC}$, the tuning region that eliminates Nadir through DC is r_r^{-1} such that

$$r_{\rm r}^{-1} \le m \left(\tau^{-1} - 2\sqrt{\tau^{-1} r_{\rm t}^{-1}/m} \right) - d$$
 (34)

• for $T_{\omega p, VI}$, the tuning region that eliminates Nadir through VI is (r_r^{-1}, m_v) such that

$$r_{\rm r}^{-1} \le (m + m_{\rm v}) \left(\tau^{-1} - 2\sqrt{\tau^{-1} r_{\rm t}^{-1} / (m + m_{\rm v})} \right) - d.$$
(35)

Proof: We start by deriving the Nadir elimination condition for VI. The system frequency of $\hat{T}_{\omega p, VI}$ is given by [15]

$$\bar{\omega}_{\mathrm{VI}}(t) = \frac{\sum_{i=1}^{n} u_{0,i}}{\sum_{i=1}^{n} f_i} p_{\mathrm{u,VI}}(t)$$

where $p_{u,VI}(t)$ is the unit-step response of $\hat{h}_{p,1,T,VI}(s)$. Clearly, as long as $p_{u,VI}(t)$ has no Nadir, neither does $\bar{\omega}_{VI}(t)$. Thus, as shown later, the core is to apply Theorem 4 to $\hat{h}_{p,1,T,VI}(s)$. Substituting (14) and (16) to (19) yields

$$\begin{split} \hat{h}_{\rm p,1,T,VI}(s) &= \frac{1}{\check{m}} \frac{s + \tau^{-1}}{s^2 + 2\xi \omega_{\rm n} s + \omega_{\rm n}^2} \\ \text{where} \quad \omega_{\rm n} := \sqrt{\frac{\check{d} + r_{\rm t}^{-1}}{\check{m}\tau}}, \quad \xi := \frac{\tau^{-1} + \check{d}/\check{m}}{2\sqrt{(\check{d} + r_{\rm t}^{-1})/(\check{m}\tau)}}. \quad \text{Now} \end{split}$$

we are ready to search the Nadir elimination tuning region by means of Theorem 4. An easy computation shows the following inequality: $2\xi\omega_n - \tau^{-1} = \check{d}/\check{m} < (\check{d} + r_t^{-1})/\check{m} = \omega_n^2 \tau$. Equivalently, it holds that $\xi < [1/(\omega_n \tau) + \omega_n \tau]/2$, which indicates that the second set of conditions in (26) cannot be satisfied. Hence, we turn to the first set of conditions in (26), which holds if and only if $\xi \ge 1$ and $\xi\omega_n \le \tau^{-1}$. Via simple algebraic computations, this is equivalent to

$$\tau \check{d}^2/\check{m} - 2\check{d} + \tau^{-1}\check{m} - 4r_{\rm t}^{-1} \ge 0 \text{ and } \check{d}/\check{m} \le \tau^{-1}.$$
 (36)

The first condition in (36) can be viewed as a quadratic inequality with respect to \check{d} , which holds if and only if

$$\check{d} \leq \check{m} \left(\tau^{-1} - 2\sqrt{\frac{r_{\mathrm{t}}^{-1}}{\check{m}\tau}} \right) \quad \text{or} \quad \check{d} \geq \check{m} \left(\tau^{-1} + 2\sqrt{\frac{r_{\mathrm{t}}^{-1}}{\check{m}\tau}} \right) \,.$$

However, only the former region satisfies the second condition in (36). This concludes the proof of the second statement. The first statement follows trivially by setting $m_v = 0$.

Important inferences can be made from Theorem 6. The fact that a small m tends to make the term on the right-hand side of (34) negative implies that in a low-inertia power system, it is impossible to eliminate Nadir using only DC. Undoubtedly, the

addition of m_v makes the tuning region in (35) more accessible, which indicates that VI can help a low-inertia power system improve Nadir.

We end this section by summarizing the pros and cons of each controller.

- Droop control: With only one parameter r_r⁻¹, DC can neither reduce frequency variance or synchronization cost without affecting steady-state effort share. Moreover, for low-inertia systems, DC cannot eliminate Nadir.
- Virtual inertia: VI can use its additional dynamic parameter m_v to eliminate system Nadir and relatively improve synchronization cost. However, this comes at the price of introducing large frequency variance in response to noise, and cannot be decoupled from increases in the steady-state effort share.

V. DYNAM-I-C DROOP CONTROL (IDROOP)

We now show how, by moving away from the broadly proposed approach of mimicking generators response, one can overcome the weaknesses presented in the previous section. With this aim, we introduce an alternative iDroop controller that uses dynamic feedback to make a trade-off among the several different objectives described in Section II-B. The proposed solution is described below.

Inverter Dynamics 3. (Dynamic Droop Control): The dynamics of an inverter with iDroop is given by the transfer function

$$\hat{c}_i(s) = -\frac{\nu_i s + \delta_i r_{\mathrm{r},i}^{-1}}{s + \delta_i} \tag{37}$$

where $\delta_i > 0$ and $\nu_i > 0$ are tunable parameters.

Similarly to (13) and (14), one can define a representative iDroop inverter controller as

$$\hat{c}_{\rm o}(s) = -\frac{\nu s + \delta r_{\rm r}^{-1}}{s + \delta} \tag{38}$$

with $\nu_i = f_i \nu$, $r_{r,i} = r_r / f_i$, and $\delta_i = \delta$.

In the rest of this section, we expose iDroop to the same performance analysis done for DC and VI in Section IV.

A. Steady-State Effort Share

We can show that iDroop is able to preserve the steady-state behavior given by DC and VI.

Corollary 7. (Synchronous frequency under iDroop): Let Assumption 3 hold. If $q_{r,i}$ is under the control law (37), then the steady-state frequency deviation of the system $\hat{T}_{\omega p,iDroop}$ synchronizes to the synchronous frequency given by (29).

Proof: The result follows directly from Lemma 1.

Corollary 8. (Steady-state effort share of iDroop): Let Assumption 3 hold. If $q_{r,i}$ is under the control law (37), then the steady-state effort share of the system $\hat{T}_{\omega p,iDroop}$ is given by (30).

Proof: The result follows from Theorem 1 applied to (37).

Corollaries 7 and 8 suggest that iDroop achieves the same synchronous frequency and steady-state effort share as DC and VI do, which depend on $r_{r,i}^{-1}$. Note that besides $r_{r,i}^{-1}$, iDroop provides us with two more degrees of freedom by δ_i and ν_i .

B. Power Fluctuations and Measurement Noise

The next theorem quantifies the frequency variance under iDroop through the squared \mathcal{H}_2 norm of the system $\hat{T}_{\omega dn,iDroop}$.

Corollary 9. (Frequency variance under iDroop): Let Assumptions 1 and 2 hold. The squared \mathcal{H}_2 norm of $\hat{T}_{\omega dn,iDroop}$ is given by

$$\begin{aligned} \|\hat{T}_{\omega \mathrm{dn},\mathrm{iDroop}}\|_{\mathcal{H}_{2}}^{2} \\ &= \sum_{k=1}^{n} \Gamma_{kk} \frac{(\kappa_{\mathrm{p}}^{2} + r_{\mathrm{r}}^{-2} \kappa_{\omega}^{2}) m \delta^{2} + (\kappa_{\mathrm{p}}^{2} + \nu^{2} \kappa_{\omega}^{2}) \left(\check{d}\delta + \lambda_{k}\right)}{2m \left[\check{d}m\delta^{2} + (d+\nu) \left(\check{d}\delta + \lambda_{k}\right)\right]}. \end{aligned}$$
(39)

Proof: The proof is based on the Theorem 2 and Lemma 2. Applying (13) and (38) to (19) and (20) shows $\hat{h}_{p,k,iDroop}(s)$ is a transfer function with $b_4 = a_0 = b_0 = 0, a_1 = (\lambda_k \delta)/m, b_1 = 0, a_2 = (\check{d}\delta + \lambda_k)/m, b_2 = \delta/m, a_3 = (m\delta + d + \nu)/m, b_3 = 1/m$, while $\hat{h}_{\omega,k,iDroop}(s)$ is a transfer function with $b_4 = a_0 = b_0 = 0, a_1 = (\lambda_k \delta)/m, b_1 = 0, a_2 = (\check{d}\delta + \lambda_k)/m, b_2 = -(r_r^{-1}\delta)/m, a_3 = (m\delta + d + \nu)/m, b_3 = -\nu/m$. Thus, by Lemma 2,

$$\begin{aligned} \|\hat{h}_{\mathrm{p},k,\mathrm{iDroop}}\|_{\mathcal{H}_{2}}^{2} &= \frac{m\delta^{2} + d\delta + \lambda_{k}}{2m\left[\check{d}m\delta^{2} + (d+\nu)\left(\check{d}\delta + \lambda_{k}\right)\right]} \\ \|\hat{h}_{\omega,k,\mathrm{iDroop}}\|_{\mathcal{H}_{2}}^{2} &= \frac{r_{\mathrm{r}}^{-2}m\delta^{2} + \nu^{2}\left(\check{d}\delta + \lambda_{k}\right)}{2m\left[\check{d}m\delta^{2} + (d+\nu)\left(\check{d}\delta + \lambda_{k}\right)\right]}. \end{aligned}$$

Then, (39) follows from Theorem 2.

The explicit expression of $\|\hat{T}_{\omega dn,iDroop}\|_{\mathcal{H}_2}^2$ given in Corollary 9 is useful to show that iDroop can reduce the frequency variance relative to DC and VI. Given the fact that $\|\hat{T}_{\omega dn,VI}\|_{\mathcal{H}_2}^2$ is infinite, the question indeed lies in whether we can find a set of values for parameters δ and ν that ensure $\|\hat{T}_{\omega dn,iDroop}\|_{\mathcal{H}_2}^2 \leq$ $\|\hat{T}_{\omega dn,DC}\|_{\mathcal{H}_2}^2$. Fortunately, we can not only find such a set but also the optimal setting for (39). The following three lemmas set the foundation of this important result which is given as Theorem 7.

Lemma 5. (Limit of $\|\hat{T}_{\omega dn,iDroop}\|_{\mathcal{H}_2}^2$): Let Assumptions 1 and 2 hold. If $\delta \to \infty$, then $\|\hat{T}_{\omega dn,iDroop}\|_{\mathcal{H}_2}^2 = \|\hat{T}_{\omega dn,DC}\|_{\mathcal{H}_2}^2$. *Proof:* See Appendix D.

Lemma 5 shows that $\|\hat{T}_{\omega dn,iDroop}\|_{\mathcal{H}_2}^2$ asymptotically converges to $\|\hat{T}_{\omega dn,DC}\|_{\mathcal{H}_2}^2$ as $\delta \to \infty$. The next lemma shows that this convergence is monotonically from either above or below depending on the value of the parameter ν .

Lemma 6. (ν -dependent monotonicity of $\|\ddot{T}_{\omega dn,iDroop}\|^2_{\mathcal{H}_2}$ with respect to δ): Let Assumptions 1 and 2 hold. Define

$$\alpha_1(\nu) := \frac{-\check{d}\kappa_{\omega}^2\nu^2 + \left(\kappa_{\mathrm{p}}^2 + r_{\mathrm{r}}^{-2}\kappa_{\omega}^2\right)\nu + dr_{\mathrm{r}}^{-2}\kappa_{\omega}^2 - r_{\mathrm{r}}^{-1}\kappa_{\mathrm{p}}^2}{d+\nu}$$

Then

- $\|\hat{T}_{\omega dn,iDroop}\|_{\mathcal{H}_2}^2$ is a monotonically increasing or decreasing function of $\delta > 0$ if and only if $\alpha_1(\nu)$ is positive or negative, respectively.
- $||T_{\omega dn,iDroop}||^2_{\mathcal{H}_2}$ is independent of $\delta > 0$ if and only if $\alpha_1(\nu)$ is zero.

Proof: See Appendix E.

By Lemma 6, for a given ν , if $\alpha_1(\nu) < 0$, then $\|\hat{T}_{\omega dn,iDroop}\|_{\mathcal{H}_2}^2$ always decreases as δ increases. However, according to Lemma 5, even if $\delta \to \infty$, we can only obtain $\|\hat{T}_{\omega dn,iDroop}\|_{\mathcal{H}_2}^2 = \|\hat{T}_{\omega dn,DC}\|_{\mathcal{H}_2}^2$. Similarly, if $\alpha_1(\nu) = 0$, then $\|\hat{T}_{\omega dn,iDroop}\|_{\mathcal{H}_2}^2$ keeps constant as δ increases, which means whatever δ is, we will always obtain $\|\hat{T}_{\omega dn,iDroop}\|_{\mathcal{H}_2}^2 = \|\hat{T}_{\omega dn,DC}\|_{\mathcal{H}_2}^2$. Therefore, iDroop cannot outperform DC when $\alpha_1(\nu) \leq 0$. To put it another way, Lemmas 5 and 6 imply that in order to improve the frequency variance through iDroop, one needs to set ν such that $\alpha_1(\nu) > 0$ and δ as small as practically possible. The following lemma characterizes the minimizer ν^* of $\|\hat{T}_{\omega dn,iDroop}\|_{\mathcal{H}_2}^2$ when $\delta = 0$.

Lemma 7. (Minimizer ν^* of $\|\hat{T}_{\omega dn,iDroop}\|_{\mathcal{H}_2}^2$ when $\delta = 0$): Let Assumptions 1 and 2 hold. Then

$$\nu^* := \operatorname{argmin}_{\delta=0,\nu>0} \|\hat{T}_{\omega \mathrm{dn},\mathrm{iDroop}}\|_{\mathcal{H}_2}^2$$
$$= -d + \sqrt{d^2 + (\kappa_\mathrm{p}/\kappa_\omega)^2} \,. \tag{40}$$

Proof: See Appendix F.

We are now ready to prove the next theorem.

Theorem 7. ($\|\hat{T}_{\omega dn,iDroop}\|_{\mathcal{H}_2}^2$ optimal tuning): Let Assumptions 1 and 2 hold. Define ν^* as in (40). Then,

• whenever $(\kappa_p/\kappa_\omega)^2 \neq 2r_r^{-1}d + r_r^{-2}$, for any $\delta > 0$ and ν such that

$$\nu \in [\nu^*, r_r^{-1}) \text{ or } \nu \in (r_r^{-1}, \nu^*]$$
 (41)

iDroop outperforms DC in terms of frequency variance, i.e.,

$$\|\hat{T}_{\omega dn,iDroop}\|_{\mathcal{H}_2}^2 < \|\hat{T}_{\omega dn,DC}\|_{\mathcal{H}_2}^2.$$

Moreover, the global minimum of $\|\hat{T}_{\omega dn,iDroop}\|_{\mathcal{H}_2}^2$ is obtained by setting $\delta \to 0$ and $\nu \to \nu^*$.

• if $(\kappa_p/\kappa_\omega)^2 = 2r_r^{-1}d + r_r^{-2}$, then for any $\delta > 0$, by setting $\nu \to \nu^* = r_r^{-1}$, iDroop matches DC in terms of frequency variance, i.e.,

$$\|\hat{T}_{\omega \mathrm{dn},\mathrm{iDroop}}\|_{\mathcal{H}_2}^2 = \|\hat{T}_{\omega \mathrm{dn},\mathrm{DC}}\|_{\mathcal{H}_2}^2$$

Proof: As discussed before, to guarantee $\|\hat{T}_{\omega dn,iDroop}\|_{\mathcal{H}_2}^2 < \|\hat{T}_{\omega dn,DC}\|_{\mathcal{H}_2}^2$, one requires to set ν such that $\alpha_1(\nu) > 0$. In this case, $\|\hat{T}_{\omega dn,iDroop}\|_{\mathcal{H}_2}^2$ always increases as δ increases, so choosing δ arbitrarily small is optimal for any fixed ν .

We now look for the values of ν that satisfy the requirement $\alpha_1(\nu) > 0$. Since the denominator of $\alpha_1(\nu)$ is always positive, the sign of $\alpha_1(\nu)$ only depends on its numerator. Denote the numerator of $\alpha_1(\nu)$ as $N_{\alpha_1}(\nu)$. Clearly, $N_{\alpha_1}(\nu)$ is a univariate quadratic function in ν , whose roots are as follows: $\nu_1 = r_r^{-1}$ and $\nu_2 = [(\kappa_p/\kappa_\omega)^2 - r_r^{-1}d]/d$. Provided that the highest order coefficient of $N_{\alpha_1}(\nu)$ is negative, the graph of $N_{\alpha_1}(\nu)$ is a parabola that opens downwards. Therefore, if $\nu_1 < \nu_2$, then $\nu \in (\nu_1, \nu_2)$ guarantees $\alpha_1(\nu) > 0$; if $\nu_1 > \nu_2$, then $\nu \in (\nu_2, \nu_1) \cap (0, \infty)$ guarantees $\alpha_1(\nu) > 0$. Notably, if $\nu_1 = \nu_2$, there is no feasible point of ν to make $\alpha_1(\nu) > 0$.

The condition $\nu_1 = \nu_2$ happens only if $(\kappa_p/\kappa_\omega)^2 = 2r_r^{-1}d + r_r^{-2}$, from which it follows that $\nu^* = r_r^{-1} = \nu_1 = \nu_2$. Then $\alpha_1(\nu^*) = \alpha_1(r_r^{-1}) = 0$. Therefore, by setting $\nu \to \nu^* = r_r^{-1}$, we get $\|\hat{T}_{\omega dn, Droop}\|_{\mathcal{H}_2}^2 = \|\hat{T}_{\omega dn, DC}\|_{\mathcal{H}_2}^2$. This concludes the proof of the second part.

We now focus on the case where the set $S = (\nu_1, \nu_2) \cup \{(\nu_2, \nu_1) \cap (0, \infty)\}$ is nonempty. Recall from the proof of Lemma 6 that $\|\hat{T}_{\omega dn, iDroop}\|_{\mathcal{H}_2} = \Pi(\delta, \nu)$. For any fixed $\nu \in S$, it holds that $\alpha_1(\nu) > 0$ and thus $\Pi(\delta, \nu) > \Pi(0, \nu)$ for any $\delta > 0$. Recall from the proof of Lemma 7 that ν^* is the minimizer of $\Pi(0, \nu)$. Hence, $(0, \nu^*)$ globally minimizes $\Pi(\delta, \nu)$ as long as $\nu^* \in S$. In fact, we will show next that ν^* is always within S whenever $S \neq \emptyset$.

First, we consider the case when $\nu_1 < \nu_2$, which implies that $(\kappa_p/\kappa_\omega)^2 > 2r_r^{-1}d + r_r^{-2}$. Then, we have $\nu^* > -d + \sqrt{d^2 + 2r_r^{-1}d + r_r^{-2}} = r_r^{-1} = \nu_1$. We also want to show $\nu^* < \nu_2$ which holds if and only if

$$\sqrt{d^2 + (\kappa_{\rm p}/\kappa_{\omega})^2} < \frac{(\kappa_{\rm p}/\kappa_{\omega})^2 - r_{\rm r}^{-1}d}{\check{d}} + d = \frac{(\kappa_{\rm p}/\kappa_{\omega})^2 + d^2}{\check{d}}$$

which is equivalent to $1 < \sqrt{d^2 + (\kappa_{\rm p}/\kappa_{\omega})^2}/d$. This always holds since $(\kappa_{\rm p}/\kappa_{\omega})^2 > 2r_{\rm r}^{-1}d + r_{\rm r}^{-2}$. Thus, $\nu_1 < \nu^* < \nu_2$. Similarly, we can prove that in the case when $\nu_1 > \nu_2$, $\nu_2 < \nu^* < \nu_1$ holds and thus $\nu^* \in (\nu_2, \nu_1) \cap (0, \infty)$. It follows that $(0, \nu^*)$ is the global minimizer of $\Pi(\delta, \nu)$.

Finally, by Lemma 5, $\|\hat{T}_{\omega dn, DC}\|_{\mathcal{H}_2}^2 = \Pi(\infty, \nu)$. The condition (41) actually guarantees $\nu \in S$ and thus $\alpha_1(\nu) > 0$. Then, by Lemma 6, we have $\|\hat{T}_{\omega dn, DC}\|_{\mathcal{H}_2}^2 = \Pi(\infty, \nu) > \Pi(\delta, \nu)$. This concludes the proof of the first part.

Theorem 7 shows that, to optimally improve the frequency variance, iDroop needs to first set δ arbitrarily close to zero. Interestingly, this implies that the transfer function $\hat{c}_{\rm o}(s) \approx -\nu$ except for $\hat{c}_{\rm o}(0) = -r_{\rm r}^{-1}$. In other words, iDroop uses its first-order lead/lag property to effectively decouple the dc gain $\hat{c}_{\rm o}(0)$ from the gain at all the other frequencies such that $\hat{c}_{\rm o}(j\omega) \approx -\nu$. This decouple is particularly easy to understand in two special regimes: i) If $\kappa_{\rm p} \ll \kappa_{\omega}$, the system is dominated by measurement noise and, therefore, $\nu^* \approx 0 < r_{\rm r}^{-1}$, which makes iDroop a lag compensator. Thus, by using lag compensation (setting $\nu < r_{\rm r}^{-1}$), iDroop can attenuate frequency noise; ii) If $\kappa_{\rm p} \gg \kappa_{\omega}$, the system is dominated by power fluctuations and therefore $\nu^* \approx \kappa_{\rm p}/\kappa_{\omega} > r_{\rm r}^{-1}$ which makes iDroop a lead compensator. Thus, by using lead compensation (setting $\nu > r_{\rm r}^{-1}$), iDroop can attenuate frequency noise; ii) If $\kappa_{\rm p} \gg \kappa_{\omega}$.

C. Synchronization Cost

Theorem 3 implies that the bounds on the synchronization cost of $\hat{T}_{\omega p,iDroop}$ are closely related to $\|\hat{h}_{u,k,iDroop}\|_{\mathcal{H}_2}^2$. If we can find a tuning that forces $\|\hat{h}_{u,k,iDroop}\|_{\mathcal{H}_2}^2$ to be zero, then both lower and upper bounds on the synchronization cost converge to zero. Then, the zero synchronization cost is achieved naturally. The next theorem addresses this problem.

Theorem 8. (Zero synchronization cost tuning of iDroop): Let Assumptions 1 and 3 hold. Then, a zero synchronization cost of the system $\hat{T}_{\omega p,iDroop}$, i.e., $\|\tilde{\omega}_{iDroop}\|_2^2 = 0$, can be achieved by setting $\delta \to 0$ and $\nu \to \infty$.

Proof: Since the key is to show that $\|\hat{h}_{u,k,iDroop}\|_{\mathcal{H}_2}^2 \to 0$ as $\delta \to 0$ and $\nu \to \infty$, we can use Lemma 2. Applying (14) and (38) to (19) shows $\hat{h}_{u,k,iDroop}(s) = \hat{h}_{p,k+1,T,iDroop}(s)/s$ is a transfer function with

$$a_0 = \frac{\lambda_{k+1}\delta}{m\tau}$$
, $b_0 = \frac{\delta}{m\tau}$,

$$a_{1} = \frac{\delta(\check{d} + r_{t}^{-1} + \lambda_{k+1}\tau) + \lambda_{k+1}}{m\tau}, \qquad b_{1} = \frac{\delta\tau + 1}{m\tau}$$

$$a_{2} = \frac{\delta(m + \check{d}\tau) + d + r_{t}^{-1} + \lambda_{k+1}\tau + \nu}{m\tau}, \qquad b_{2} = \frac{1}{m}$$

$$a_{3} = \frac{m\delta\tau + m + d\tau + \nu\tau}{m\tau}, \qquad b_{3} = 0, \qquad b_{4} = 0.$$

Considering that $a_0 \to 0$ and $b_0 \to 0$ as $\delta \to 0$ and $\nu \to \infty$, we can employ the \mathcal{H}_2 norm computation formula for the third-order transfer function in Remark 6. Then

$$\lim_{\delta \to 0, \nu \to \infty} \|\dot{h}_{\mathrm{u},k,\mathrm{iDroop}}\|_{\mathcal{H}_2}^2$$
$$= \lim_{\delta \to 0, \nu \to \infty} \frac{\frac{\nu}{m} \left(\frac{1}{m\tau}\right)^2 + \frac{\lambda_{k+1}}{m\tau} \left(\frac{1}{m}\right)^2}{2\frac{\lambda_{k+1}}{m\tau} \left(\frac{\nu}{m\tau} \frac{\nu}{m} - \frac{\lambda_{k+1}}{m\tau}\right)} = 0.$$

Thus, by Theorem 3, $\|\tilde{\omega}_{iDroop}\|_2^2 = \overline{\|\tilde{\omega}_{iDroop}\|_2^2} = 0$, which forces $\|\tilde{\omega}_{iDroop}\|_2^2 = 0$.

Theorem 8 shows that unlike DC and VI that require changes on r_r^{-1} to arbitrarily reduce the synchronization cost, iDroop can achieve zero synchronization cost without affecting the steady-state effort share. Naturally, $\delta \approx 0$ may lead to slow response and $\nu \to \infty$ may hinder robustness. Thus, this result should be appreciated from the viewpoint of the additional tuning flexibility that iDroop provides.

D. Nadir

Finally, we show that with δ and ν tuned appropriately, iDroop enables the system frequency of $\hat{T}_{\omega p,iDroop}$ to evolve as a firstorder response to step power disturbances, which effectively makes Nadir disappear. The following theorem summarizes this idea.

Theorem 9. (Nadir elimination with iDroop): Let Assumptions 1 and 3 hold. By setting $\delta = \tau^{-1}$ and $\nu = r_r^{-1} + r_t^{-1}$, Nadir (12) of $\hat{T}_{\omega p,iDroop}$ disappears.

Proof: The system frequency of $T_{\omega p,iDroop}$ is given by [15]

$$\bar{\nu}_{\text{iDroop}}(t) = \frac{\sum_{i=1}^{n} u_{0,i}}{\sum_{i=1}^{n} f_i} p_{\text{u,iDroop}}(t)$$
(42)

where $p_{u,iDroop}(t)$ is the unit-step response of $\hat{h}_{p,1,T,iDroop}(s)$. If we set $\delta = \tau^{-1}$ and $\nu = r_r^{-1} + r_t^{-1}$, then (38) becomes

$$\hat{c}_{\rm o}(s) = \frac{r_{\rm t}^{-1}}{\tau s + 1} - \left(r_{\rm r}^{-1} + r_{\rm t}^{-1}\right). \tag{43}$$

Applying (14) and (43) to (19) yields

$$\hat{h}_{\mathrm{p},1,\mathrm{T},\mathrm{iDroop}}(s) = \frac{1}{ms + \check{d} + r_{\mathrm{t}}^{-1}}$$

whose unit-step response $p_{u,iDroop}(t)$ is a first-order evolution. Thus, (42) indicates that Nadir of the system frequency disappears.

VI. NUMERICAL ILLUSTRATIONS

In this section, we present simulation results that compare iDroop with DC and VI. The simulations are performed on the Icelandic Power Network taken from the Power Systems Test Case Archive [34]. Instead of the linear network model used in the analysis, the simulations are built upon a nonlinear

DC

iDroop

40

40

iDroop

20 40 60

Time t(s)

Time t(s)

60

20

20

0.02

0.01

-0.01

-0.02 60

SW

VI

40

40

0.2

0.1

-0.1

-0.2

Time t(s)

60 0

60 0

(a)

VI

Time t(s)

(b)

20

20

DC

40 60 0 20 40

Time t(s)

Frequency Deviation $\omega(Hz)$

ency Deviation ω(Hz) 70- 70- 70 70- 70

-0.6

0.02

0.01

-0.01

-0.02

of frequency deviations.

Control Effort q_r(p.u.)

0.2

0 -0.2 -0.4 -0.6

TABLE I PARAMETERS OF REPRESENTATIVE GENERATOR AND INVERTER

Parameters	Symbol	Value
generator inertia	m	$0.0111{ m s}^2{ m rad}^{-1}$
generator damping	d	$0.0003{ m srad^{-1}}$
turbine time const.	τ	$4.59\mathrm{s}$
turbine droop	$r_{ m t}$	374.49 rad s ⁻¹ for SW 748.97 rad s ⁻¹ otherwise
inverter droop	$r_{ m r}$	$748.97 \mathrm{rad s^{-1}}$

setup including voltage dynamics and nonlinear power flows. The dynamic model contains 35 generator buses and 83 load buses. Each of generator buses for simplicity is distinctly indexed by some $i \in \{1, \ldots, 35\}$ here. Even though our previous analysis requires the proportionality assumption (Assumption 1), in the simulations, for every generator bus i, the generator inertia coefficient, the turbine time constant, and the turbine droop coefficient are directly obtained from the dataset, i.e., $m_i = m_{d,i}, \tau_i = \tau_{d,i}$, and $r_{t,i} = r_{t,d,i}$.^{6,7} In addition, turbine governor deadbands are taken into account such that turbines are only responsive to frequency deviations exceeding ± 0.036 Hz. Given that the values of generator damping coefficients are not provided by the dataset, we set $d_i = f_i d$ with d being the representative generator damping coefficient and $f_i := m_i/m$ being the proportionality parameters, where m is the representative generator inertia defined as the mean of m_i 's, i.e., $m := (\sum_{i=1}^{n} m_i)/n$. For every load bus, the damping coefficient is chosen as the mean of all generator damping coefficients. We refer to this system without inverter control as "SW" in the simulations.

We then add an inverter to each generator bus *i*, whose control law is either one of DC, (filtered) VI, and iDroop.⁸ The design of controller parameters will be based on the representative generator parameters. Hence, besides *m* and *d*, we define $\tau :=$ $(\sum_{i=1}^{n} \tau_{d,i})/n$ and $r_t := (\sum_{i=1}^{n} f_i)(\sum_{i=1}^{n} r_{t,d,i}^{-1})$. Note that to keep the synchronous frequency unchanged, once inverters are added, we halve the inverse turbine droop $r_{t,i}^{-1}$ and assign the representative inverter droop coefficient r_r a value such that the inverse inverter droop $r_{r,i}^{-1} := f_i r_r^{-1}$ should exactly compensate this decreased r_t^{-1} in the absence of turbine governor deadbands. The values of all the representative parameters mentioned above are given in Table I.

Although our current theoretical analysis does not contemplate jointly the effects of step and stochastic disturbances, we illustrate here that the Nadir-eliminated tuning of Theorem 9 for iDroop can perform quite well in more realistic scenarios with combined step and stochastic disturbances. In Fig. 4, we show how different controllers perform when the system is subject to a step drop of -0.3 p.u. in power injection at bus



number 2 at time t = 1s as well as power fluctuations and measurement noise. Since in reality power fluctuations are larger than measurement noise, we focus on the case dominated by power fluctuations, where $\kappa_{\rm p} = 10^{-3}$ and $\kappa_{\omega} = 10^{-4}$. As for the representative inverter, we turn $\delta = \tau^{-1} = 0.218 {\rm s}^{-1}$ and $\nu = r_{\rm r}^{-1} + r_{\rm t}^{-1} = 0.004 {\rm srad}^{-1}$ in iDroop such that Nadir of the system frequency disappears as suggested by Theorem 9 and we tune $m_{\rm v} = 0.0445 s^2 r a d^{-1}$ in VI such that the system frequency

⁶Throughout this section, we use the subscript d, *i* to denote the original parameters of the *i*th generator bus from the dataset.

⁷For illustrative purpose only, we reassign a part of the droop $r_{t,d,i}$'s on turbines in the dataset to let there be a deeper Nadir in the system frequency.

⁸We add a low-pass filter with time constant 0.0001s to VI in simulation since it is required in reality for a controller with a derivative term.

is critically damped. The inverter parameters on each bus *i* are defined as follows: $\delta_i := \delta$, $\nu_i := f_i \nu$, and $m_{v,i} = f_i m_v$.

Some observations are in order. First, all three controllers lead to the same synchronous frequency as predicted by Corollaries 1 and 7. Second, although both of VI and iDroop succeed in eliminating Nadir of the system frequency-which is better than what DC does-the system synchronizes with faster rate and lower synchronization cost under iDroop than VI. Interestingly, although our simulation violates the homogeneous assumption, the synchronization costs under VI, DC, and SW are in ascending order, which aligns with Corollary 5. Third, there is no surprise that the frequency variance under VI is no longer unbounded due to the added filter, yet, for a filter with a bandwidth that does not affect the system dynamics, the system under VI is still more sensitive to noise compared to the one under DC or iDroop as illustrated by the empirical PDF. Moreover, we highlight the huge control effort required by VI when compared with DC and iDroop. Last but not least, a bonus of the Nadir-eliminated tuning is that iDroop outperforms DC in frequency variance as well. This can be understood through Theorem 7. Provided that $\kappa_p \gg \kappa_\omega$, we know from the definition in Lemma 7 that $\nu^* \approx \kappa_p / \kappa_\omega$. Thus, for realistic values of system parameters, $\nu^* \gg r_r^{-1}$ always holds. It follows directly that $\nu = r_r^{-1} + r_t^{-1} \in (r_r^{-1}, \nu^*]$. By Theorem 7, iDroop performs better than DC in terms of frequency variance. Further, the preceding simulation results suggest that the Nadir-eliminated tuning of iDroop designed based on the proportional parameters assumption works relatively well even when parameters are non-proportional.

VII. CONCLUSION

This article studies the effect of grid-connected inverter-based control on the power system performance. When it comes to the existing two common control strategies, we show that DC cannot decouple the dynamic performance improvement from the steady-state effort share and VI can introduce unbounded frequency variance. Thus, we propose a new control strategy named iDroop, which is able to enhance the dynamic performance and preserve the steady-state effort share at the same time. We show that iDroop can be tuned to achieve strong noise rejection, zero synchronization cost, and frequency Nadir elimination when the system parameters satisfy the proportionality assumption. We illustrate numerically that the Nadir-eliminated tuning designed based on the proportional parameters assumption strikes a good trade-off among various performance metrics even if parameters are nonproportional. This article illustrates the superiority of principled control design when compared with naive approaches such as VI.

APPENDIX

A. Proof of Lemma 1

Combining (1) and (7) through $u_{\rm P} = p_{\rm in} - p_{\rm e}$, we get the (partial) state-space representation of the system $\hat{T}_{\omega \rm p}$ as

$$\theta = \omega$$
, (44a)

$$M\dot{\omega} = -D\omega - L_{\rm B}\theta + q_{\rm r} + q_{\rm t} + p_{\rm in} \tag{44b}$$

where $M := \operatorname{diag}(m_i, i \in \mathcal{V}) \in \mathbb{R}_{\geq 0}^{n \times n}$, $D := \operatorname{diag}(d_i, i \in \mathcal{V}) \in \mathbb{R}_{\geq 0}^{n \times n}$, $q_r := (q_{r,i}, i \in \mathcal{V}) \in \mathbb{R}^n$, and $q_t := (q_{t,i}, i \in \mathcal{V}) \in \mathbb{R}^n$. In steady state, (44) yields

$$L_{\rm B}\omega_{\rm ss}t = -D\omega_{\rm ss} - L_{\rm B}\theta_{\rm ss_0} + q_{\rm r,ss} + q_{\rm t,ss} + u_0 \tag{45}$$

where $(\theta_{ss_0} + \omega_{ss}t, \omega_{ss}, q_{r,ss}, q_{t,ss})$ denotes the steady-state solution of (44). Equation (45) indicates that $L_B\omega_{ss}t$ is constant and thus $L_B\omega_{ss} = \mathbb{O}_n$. It follows that $\omega_{ss} = \omega_{syn}\mathbb{1}_n$. Therefore, (45) becomes

$$\mathbb{O}_n = -D\omega_{\rm syn}\mathbb{1}_n - L_{\rm B}\theta_{\rm ss_0} + q_{\rm r,ss} + q_{\rm t,ss} + u_0 \tag{46}$$

where $q_{r,ss} = (\hat{c}_i(0)\omega_{syn}, i \in \mathcal{V}) \in \mathbb{R}^n$ and $q_{t,ss} = (-r_{t,i}^{-1}\omega_{syn}, i \in \mathcal{V}) \in \mathbb{R}^n$ when $\omega_{\epsilon} = 0$ by (2). Premultiplying (46) by $\mathbb{1}_n^T$ and using the property that $\mathbb{1}_n^T L_B = \mathbb{0}_n^T$, we get the desired result in (22).

B. Proof of Lemma 2

First recall that given any state-space realization of $\hat{h}(s)$, the \mathcal{H}_2 norm can be calculated by solving a particular Lyapunov equation. More specifically, suppose

$$\Sigma_{\hat{h}(s)} = \begin{bmatrix} A & B \\ \hline C & D \end{bmatrix}$$

and let X denote the solution to the Lyapunov equation

$$AX + XA^T = -BB^T. ag{47}$$

If $\hat{h}(s)$ is stable, then

$$\|\hat{h}\|_{\mathcal{H}_2}^2 = \begin{cases} \infty & \text{if } D \neq 0\\ CXC^T & \text{otherwise.} \end{cases}$$
(48)

Consider the observable canonical form [35, Ch. 4.1.6] of $\hat{h}(s)$ given by

$$\Sigma_{\hat{h}(s)} = \begin{bmatrix} 0 & 0 & 0 & -a_0 & b_0 \\ 1 & 0 & 0 & -a_1 & b_1 \\ 0 & 1 & 0 & -a_2 & b_2 \\ 0 & 0 & 1 & -a_3 & b_3 \\ \hline 0 & 0 & 0 & 1 & b_4 \end{bmatrix} .$$
(49)

Since $D = b_4$, it is trivial to see from (48) that if $b_4 \neq 0$, then $\|\hat{h}\|_{\mathcal{H}_2}^2 = \infty$. Hence, in the rest of the proof, we assume $b_4 = 0$. We will now solve the Lyapunov equation analytically for the realization (49). X must be symmetric and thus can be parameterized as

$$X = [x_{ij}] \in \mathbb{R}^{4 \times 4}, \quad \text{with} \quad x_{ij} = x_{ji}. \tag{50}$$

Since it is easy to see that $CXC^T = x_{44}$, the problem becomes solving for x_{44} . Substituting (49) and (50) into (47) yields the following equations:

$$2a_0 x_{14} = b_0^2 \tag{51a}$$

$$x_{12} - a_2 x_{14} - a_0 x_{34} = -b_0 b_2 \tag{51b}$$

$$2(x_{12} - a_1 x_{24}) = -b_1^2 \tag{51c}$$

$$x_{23} - a_3 x_{24} + x_{14} - a_1 x_{44} = -b_1 b_3 \tag{51d}$$

$$2(x_{23} - a_2 x_{34}) = -b_2^2 \tag{51e}$$

$$2(x_{34} - a_3 x_{44}) = -b_3^2 .$$
 (51f)

Through standard algebra, we can solve for x_{44} , which gives the right hand side of (24) the denominator is guaranteed to be nonzero by the Routh–Hurwitz criterion. This concludes the proof.

C. Proof of Lemma 4

First note that

$$\begin{aligned} x^{T}\left(P\circ\left(yy^{T}\right)\right)x &= \operatorname{tr}\left(P^{T}\left(x\circ y\right)\left(x\circ y\right)^{T}\right) \\ &= \left(x\circ y\right)^{T}P^{T}\left(x\circ y\right). \end{aligned}$$

Let $w := x \circ y$. Since P is symmetric, by Rayleigh [31]

 $\lambda_{\min}(P)w^T w \le x^T \left(P \circ \left(yy^T \right) \right) x \le \lambda_{\max}(P)w^T w.$

Observing that $w^T w = \sum_{k=1}^n x_k^2 y_k^2$ completes the proof.

D. Proof of Lemma 5

The limit of (39) as $\delta \to \infty$ can be computed as

$$\lim_{\delta \to \infty} \|\hat{T}_{\omega \mathrm{dn,iDroop}}\|_{\mathcal{H}_2}^2 = \sum_{k=1}^n \Gamma_{kk} \frac{\kappa_{\mathrm{p}}^2 + r_{\mathrm{r}}^{-2} \kappa_{\omega}^2}{2m\check{d}} = \|\hat{T}_{\omega \mathrm{dn,DC}}\|_{\mathcal{H}_2}^2$$

where the second equality follows from (31a).

E. Proof of Lemma 6

Provided that $\|\hat{T}_{\omega dn,iDroop}\|_{\mathcal{H}_2}^2$ is a function of δ and ν , in what follows, we denote it by $\Pi(\delta,\nu)$. To make it clear how $\Pi(\delta,\nu)$ changes with δ , we first put it into the equivalent form of

$$\Pi(\delta,\nu) = \sum_{k=1}^{n} \Gamma_{kk} \left[\frac{\alpha_1(\nu)\delta^2}{\alpha_2\delta^2 + \alpha_3(\nu)\delta + \alpha_4(\nu,\lambda_k)} + \alpha_5(\nu) \right]$$

with

$$\begin{aligned} \alpha_1(\nu) &:= \frac{-\check{d}\kappa_{\omega}^2\nu^2 + \left(\kappa_{\rm p}^2 + r_{\rm r}^{-2}\kappa_{\omega}^2\right)\nu + dr_{\rm r}^{-2}\kappa_{\omega}^2 - r_{\rm r}^{-1}\kappa_{\rm r}^2}{d+\nu} \\ \alpha_2 &:= 2m\check{d} \;, \qquad \alpha_3(\nu) := 2(d+\nu)\check{d} \\ \alpha_4(\nu,\lambda_k) &:= 2(d+\nu)\lambda_k \;, \qquad \alpha_5(\nu) := \frac{\kappa_{\rm p}^2 + \nu^2\kappa_{\omega}^2}{2m(d+\nu)} \;. \end{aligned}$$

We then take the partial derivative of $\Pi(\delta, \nu)$ with respect to δ as

$$\partial_{\delta} \Pi(\delta, \nu) = \alpha_1(\nu) \sum_{k=1}^n \Gamma_{kk} \left[\frac{\alpha_3(\nu)\delta^2 + 2\alpha_4(\nu, \lambda_k)\delta}{(\alpha_2\delta^2 + \alpha_3(\nu)\delta + \alpha_4(\nu, \lambda_k))^2} \right]$$

Since m > 0, d > 0, $\nu > 0$, and $r_r^{-1} > 0$, α_2 and $\alpha_3(\nu)$ are positive. Also, given that all the eigenvalues of the scaled Laplacian matrix L_F are non-negative, $\alpha_4(\nu, \lambda_k)$ must be nonnegative. Thus, $\forall \delta > 0$, $(\alpha_3(\nu)\delta^2 + 2\alpha_4(\nu, \lambda_k)\delta)/(\alpha_2\delta^2 + \alpha_3(\nu)\delta + \alpha_4(\nu, \lambda_k))^2 > 0$.

Recall from the proof of Corollary 4 that $\Gamma_{kk} > 0$, $\forall k \in \{1, ..., n\}$. Therefore, $\forall \delta > 0$, $\operatorname{sign}(\partial_{\delta} \Pi(\delta, \nu)) = \operatorname{sign}(\alpha_1(\nu))$.

F. Proof of Lemma 7

Recall from the proof of Lemma 6 that $\|\ddot{T}_{\omega dn,iDroop}\|_{\mathcal{H}_2}^2 = \Pi(\delta,\nu)$. Then, we have

$$\Pi(0,\nu) = \frac{\kappa_{\rm p}^2 + \nu^2 \kappa_{\omega}^2}{2m(d+\nu)} \sum_{k=1}^n \Gamma_{kk}$$

whose derivative with respect to ν is given by

$$\Pi'(0,\nu) = \frac{\kappa_{\omega}^2 \nu^2 + 2d\kappa_{\omega}^2 \nu - \kappa_{\rm p}^2}{2m(d+\nu)^2} \sum_{k=1}^n \Gamma_{kk} .$$
 (52)

Note that (52) and (33) are in the same form. Thus, ν^* is determined in the same way as in the proof of Corollary 4.

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